N552TC

EPILATUS Z



Book	No.	01

AIRCRAFT LOG =PILATUS=

CH-6371 STANS SWITZERLAND

PC-12

443

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AIRCRAFT DESCRIPTION

Registration No. N 466 SA

Model:	PC-12 /45	
Serial No.:	443	Type Certificate No.: A 78 EU
Colors:		
Engine Manufacturer:	PRATT + WHITNEY	Propeller Manufacturer: HAR 12ELL
Original Engine Mode	PTGA- 67B	Original Propeller Model: #C-E44-3D /
		E 10477K
Serial Number:	PCE - PR 0309	Hub or Serial Number:

Blade Serial Number:

J-64500 J-64502 J-64499 J-64513

PROPELLER INFORMATION

Dulo			Diam.			Serial Numb	oers	Mark York	Desta
Date Installed	Manufacturer	Model	In.	Hub	Blades		des	CORLOR	Date Removed
4 0 mm				HUB	1	2	3	4	Komovou
101 23 2001	HARTZELL	HC-E4A-30	104	HJ-1660	764500	J 64502	J64499	164513	4.50
NA	3202 2486	3 Reve A	P a Sa	HH	.c.i.noil	searly.	annaA.	ino.	25.10
		de la company							

	100-HOUR INSPECTION SUMMARY (SEE LOG FOR DETAILS)						ANNUAL INSPECTION SUMMARY (SEE LOG FOR DETAILS)	
Certificate No.	Accomplished By	Recorder Reading			Certificate No.	libutt.	Accomplished By	Date
25 Mariana 3872.5	100 An. Inspection i.a.v. 114 07049 Port		100:39	12.08.07	A+P2468693 [A	Rev. 13	Inspection ia.v. HM 02049	07.11.03
9 Rus 2468698 L	100 Hrs. Inspection i.a.u. MM (2049) Rour		275:12		ATT 24686 93 IA	Rev. 14	Inspection i.a.w. MM 02049	9.10.04
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THE CHIEF !

	COMPLIANCE WITH FAA AIRWOR	COMPLIANCE WITH FAA AIRWORTHINESS DIRECTIVES							
A. D. No. and Date	Method of Compliance (See Log for Details)	Time on Aircraft	Date of Compliance	Signature and Certificate No.	A. D. No. and Date	Method of Compliance (See Log for Details)	Time on Aircraft	Date of Compliance	Signature and Certificate No.
2001-22-18	18 57-004 compl. during product	m 0:00	Dec. 10, 2001	(2)	2004-12-11	SB 27-015	247.3	06.08.04	A+P. 2468643 ZA
	1877.008 compl. during producti		Dec. 10. 2001	(2)	2004-10-15	KPNDR N/A component not install	423:03	29.10.04	A+1/2468693.IA
2002-01-09	1878-009 compl. during production		Der. 10. 700 1	West Car	2005-01-11	513 32-017 NIA by actuator SIN	449.6	Mar. 14. 05	16. 249 846.4
2007-06-10	18 35.007 compl. during producti		Dec. 10, 200	1.11	2005-04-16	SB 30-009 NIA by SIN	449 6	Mar. 14.05	18 P 24 9 8464
2003-07-10	1831-003 Sun Jet Center	183.4	19.17, 07		1	mill mid are liminis	opi ha mil	Brand M	58.74-004
7007-14-77	1832.012 N/A by IN		01.08.03	4+9 7651109	3	MALIN A. MY LINE	Andrews	i by II MT	CR 78-072
	SB 25.025 NIA by AIC IN	247.7	05.08.03	1.8 7651109	E	on the on the same and a	were D. Jef	Les Bed	200-05 B
2003-11-17	SB 53-003 Revet N/A by A/C JN	747.3	05. 08. 03	A+ 8 765 1109	1-25	and maker like	-nissier	W. Ou	700-N 82
2003-14-07	SB 32-014 Deva (due at 4000 LDG:s)		06.08.03		1	80 M 51 SH-875	well 4	(Least State	60 P
	SB 28-011 Rev. 1	247.3	06.08.03	A+P24(8693 IA	I BY YEL	428: 05 23.40.04	A Lud or	Language No.	L. Sharles
2004-04-01	SB-28-010/SB-28-009	362.9	26.04.04	Sun Aciation		123:03 7210.04	e tide i	J. sull Mary L.	P === 22 82
2004-06-05	\$8 32-015	362.9	26.04.04	Sun Aviation	•	1 4.23 to 3 122. As. of 1	P. 2 (1907) 3,0000	A Kadry Jo	d ka dea de

CO	MPLIANCE WITH MANUFACTURER'S SER	VICE BULLETIN	S AND LETTER	,	cc	OMPLIANCE WITH MANUFACTURER'S SEI	RVICE BULLETIN	IS AND LETTER	s
Bulletin or Letter No.	Subject and Remarks	Time on Aircraft	Date of Compliance	Signature and Certificate No.	Bulletin or Letter No.	Subject and Remarks	Time on Aircraft	Date of Compliance	Signature and Certificate No.
JB 31-004 Rev	A Mod. Center Adv. Warring Just	247.3	05.08.07) Elm	5B 76 - 001	Guard on Condition Lever	423:03	29.10.04	AF 265-1060
717 58 - 044	Fred Booster Pump Damage	747.3	05.08.03	She					
5/5 79-0049/64	4 Oil Press Tromducer Repl.	247.3	05.08.03						
18 79-015	Repl. inboard outh. fex 14	of 247.3	01.01.23	fund					
	DC Gen 1 Grand serring kod		02.11.03	fue					
	DC Roph. Generalor Control Unit		07.41.03	& for	-				
5B 30-008	70 50		07.11.03	1 - "					
	Acc. Chip Detector Install.		07.41.03						
SL 077 SB 32-016	Crev Sect Pour LDG inspection and replacement of components - Part A		29.10.04	1					
SB 33-009	Dual Nav. Lights	423:03	1910.04	40 4					
SB 35-004	Oz pipes - change routing	423:03	29.10.04	Dry 3					

AIRCRAFT LOG

		AIRCK	AFT LOG			
Date	Aircra	ft Time	Recorder	M	- , -	and Remarks
Date	This Date	Total	Reading	Maintenance		did kelildiks
	Brought Forward					The aircraft components and / or Emergency Locator Transmitter (ELT) has and FAA FAR Part 91.207.
					_	ELT make is: KANNAD, ☐ Model is: 1/2 Model is: 4/2 Battery make is: SERPE-IESM, and replace
	dance wi	aft was manufac th approved pro Type Design.			- -	Automatic pressure altitude reporting eq is found to be in compliance with FOCA do
		pected this aircr	aft and issued a	Certificate of	- -	Altimeter (s) and static system has/have be be in compliance with FOCA document TN Altimeter(s) S/N(s) 4561 (b)
	dated	10. DEZ. 2001			_ , -	ATC transponder (s) has/have been tested found to be in compliance with FOCA doc
	Signed	1. July	Title	Insp.	- 1 -	Installed ATC transponder are:
		/			_ v _	2. MakeAllied Signal
al to	1	1 100	4.1	7 Ldg'6.		Authorization PILATUS

emarks	Sign	nature Certific	ate No.
The aircraft components and / or systems: Emergency Locator Transmitter (ELT) has been insta and FAA FAR Part 91.207. ELT make is: KANNAD, Model is: 121 AF Model is: 406 AF Battery make is: SERPE-IESM, and replacement dat	lled per Pilatus Aircraft Ltd. Doo 6/N is:	cument AQ-GT-12-12/79	ated:
Automatic pressure altitude reporting equipment is found to be in compliance with FOCA document T Altimeter (s) and static system has/have been tested to be in compliance with FOCA document TM-W-20.03 Altimeter(s) S/N(s) 4561 (left/pilot) a	has been tested per Pilatus Aircr M-W-20.020-20 and FAA FAR per Pilatus Aircraft Ltd. Docume 20.20 and FAA FAR Part 91.411	raft Ltd. Document AQ-CD-12- Part 91.217; and ent AQ-CD-12-014, and is/are f	found to
ATC transponder (s) has/have been tested per Pilatt found to be in compliance with FOCA document TM Installed ATC transponder are:	us Aircraft Ltd. Document AQ-C I-W-20.100-20 and FAA FAR Pa	D-12-014, and is/are urt 91.413.	
1. MakeAllied Signal Model_ 2. MakeAllied Signal Model_	KT-70 S/N 30	641 572	
1. MakeAllied SignalModel_	KT-70 S/N 30	572	-

		AIRC	RAFT LOG	
Date	Aircraf	t Time	Recorder	
Dule	This Date	Total	Reading	Maintenance
z. 10/2001	Brought		4.7	7 ldg's.
c. 13/2001				
The ne	ave issued a 57 rthiness Certifica ext inspection is that is Smith D	ate. Dated 13		
Tay 17. 200	z	100:39		67 ldg.
Total to				

Make/Model PILATUS PC-12		Reg No.: N466S
Serial No. 443		
Time: <u>ACTT: 66.1</u>	AIR SERVICE The Aviation Professionals	Date: <u>04/04/0</u>
Time. ACTT. 00.1	Log Book Entry	
THE CALL DI TERMINAL TO	CHATTLE AND NOT ALLOWING VOLTAGE THRO	UGH THE CONTACTS. REMOVE
CHECKS COMPLETED ON WORK	PERATIVE AND NOT ALLOWING VOLTAGE THRO AND INSTALLED A NEW TERMINAL BOARD P/N K DESCRIBED ABOVE AND FOUND TO BE SATIS LATUS PC-12 MAINTENANCE MANUAL AND IS A	N 971.42.31.705. OPERATIONA
CHECKS COMPLETED ON WORK ACCOMPLISHED I/A/W THE PIL	NAND INSTALLED A NEW TERMINAL BOARD P/N K DESCRIBED ABOVE AND FOUND TO BE SATTS	N 971.42.31.705. OPERATIONA
CHECKS COMPLETED ON WORI ACCOMPLISHED I/A/W THE PII SERVICE. END. Work Order: 18628	AND INSTALLED A NEW TERMINAL BOARD PIN K DESCRIBED ABOVE AND FOUND TO BE SATIS LATUS PC-12 MAINTENANCE MANUAL AND IS A	N 971.42.31.705. OPERATIONA FACTORY. ALL WORK PPROVED FOR RETURN TO

基础设置

2.45.47	PILATUSE
A/C REG # N466 JA DATE 17. 05.07 TOTAL A/C TIME 100:39 HRS. TACH TIME	AIRCRAFT MAINTENANCE CH-6371 STANS / SWITZERLAND
ACCORDANCE WITH AN (CIRCLE ONE) ANNUAL/100 HR INSPECTION AS PER MM 070 43 AND WAS	
DETERMINED TO BE IN AN AIRWORTHY CONDITION.	
12 2651060 AFP	
SIGNATURE CERTIFICATE NOTITEE (X	

R ₁ gistrat	ion No		
and Remarks	Recorder	Signature	Certificate No.
Aircra	ft Log		
Date:12/17/2002	N:466SA		
Hobbs: WO:16114	Aircraft S/N:	443	
Installed customer supposed MAINTENANCE RELEAR propeller or appliance was regulations of the Federal A All work was performed in maintenance manuals and in Signature:	repaired and inspected in Aviation Administration accordance with applications. Pertinent descriptions.	ame, aircraft engine, n accordance with current and is approved for return	to service.

Signature: ECWR470D

Sun Aviation, Inc. Municipal Airport, Vero Beach, FL 32961 (772) 562-9257

DA	TE: 12/19/2002 SUN JET	~	Page 1 of 2
M/	AKE / MODEL: Pilatus PC 12/45		AIRCRAFT REG: N433SA
AIF	RCRAFT TOTAL TIME: 183.4 AIRFRAME	LC	OG ENTRY AIRCRAFT S/N: 443
inspe	tify that this aircraft has been inspected I/A/W an annual inspection checklist, FAR 91.409(a), and has been determined to be dy 2002-25B.	ction in an	per guidelines in the Pilatus PC 12/45 maintenance manual annual airworthy condition. Airworthiness Directives verified through bi-
	worthiness Directives complied with:	Spe	cial Inspections complied with:
>	No outstanding AD's apply at this time.	>	30 Day / 12 Month Fire Extinguisher Weight Check and Inspection. No defects noted at this time. Fire extinguisher weight 4 Lb., 10.5 Oz's.
Ser	vice Information complied with:	A .	100 Hour Horizontal Stabilizer Trim Warning Test. No defects noted at this time. #1 & #2 Battery Capacitance Test and Service. No defects noted
>	SB 33-006, Cabin Flood Light Diode Installation. Installed diode P/N 980.71.10.101.	>	at this time. 200 Hour Starter Generator Brush Inspection. No defects noted at
*	SB 71-006, By repositioning of the Engine Electrical Harness.	>	this time. 200 Hour Flap Electrical Warning Test. No defects noted at this time.
>	SB 11-003, by installation of Placard P/N 511.32.12.120, SB 31-003, By installation of Switch Cover P/N 500.50.12.273.	>	One Year Flap Actuator Backlash Inspection. L/H O/B S/N 0647=,008, L/H I/B S/N 0639=,008, R/H O/B S/N 0649=,007, R/H I/B S/N 0641=,006, All satisfactory at this time.
		* *	Six month De-ice Boot Treatment. 6 / 12 Month Emergency Battery Inspection and Capacitance Check. No defects noted at this time.
FA	R 91 Compliance :	-	One Year Lubrication of all Wheel Bearings. No defects noted at this time.
>	Complied with FAR 91.207 paragraphs c & d. ELT shock test and battery check. Battery due date 01/08. FAR 91.411 & 91.413 previously accomplished 12/01. Next due 12/03.	A A	One Year Clutch Torque Check on the Pitch, Roll, and Yaw Servos. No defects noted at this time. One Year Inspection of the Main Landing Gear Bushings and Axles. No defects noted at this time.

*******Inspection Continued on the Next Page******Sig. / Initials

 Removed, lubricated, insulated, and re- environment. Serviced the oxygen bot main wheel assembly. Reassembled ar 	tle to 1850 PSL. Removed, disa and installed the wheel assembly	ssembled and installed a new ti Installed correct stainless stee	re P/N 8.50X10 10 MAV on the L/H el screws in the flux detector access				
panels. Removed and reinstalled the aircraft interior for installation of phone antenna by Sun Avionics. See 8110-3 provided by Su Avionics for details. Serviced the hydraulic reservoir to 56 PSL.							
Operational / leak checks completed or 12/45 maintenance manual and is appr	n the work described above and		work accomplished UA/W the Pilatus PC				
Maintenance Release: This aircraft, airframe, Aviation Administration and is approved for retrappliance maintenance manuals and instructions Authorized Signature: L. J. Sun Aviation, Inc. Municipal Airport,	urn to service. All work was per s. Pertinent details are on file at t Ben T, Game	formed in accordance with the ap his repair station under work orders. Chief Inspector for CRS	plicable aircraft, engine, propeller, and / or er_16002,				
		Julies cells and					

Miscellaneous maintenance performed during this inspection:

Page 2 of 2

JATURE

-		AIRC	RAFT LOG		
	Aircraft Time		Recorder		
Date	This Date	Total	Reading		Maintenance
	Brought			77	
The same)	-7 TH			
Professionals Serving		4300 S. Ken	nedy ST., BOISE, IDAHO 8	3705, U.S.A.	PHONE 208-338-1800
		ACFTLO	GBOOK INSI	ERT	
AKE: Pilatus	MODEL: PC-12/4		: N466SA	S/N: 443	HR METER: 221.9
AINTENANCE RELEA	SE: The Aircraft and or compone airworthy with respect to the under Shop Work Order No. Signature of authorized Perso	work performed, subseque W6876.	naired and or inspected, per cu inthy approved for return to se	yoh	Federal Aviation Administration, and found to be 1903. Pertinent details are on file at this agency Dennis A. Snyder
	4-140-1-1				
			- 0 m		
Total to Carry Forward	A CONTRACTOR OF THE PARTY OF TH				

SUN JET CENTER

AIRCRAFT REG: N466SA

AIRCRAFT S/N: 443

AIRCRAFT TOTAL TIME: 202.8 AIRFRAME LOG ENTRY

DATE: 03/31/2003

MAKE / MODEL: Pilatus PC12/45

- Removed the L/H main strut, disassembled, cleaned, inspected and reassembled utilizing strut kit # 532.10.12.500. Serviced and reinstalled the strut.
- Leak checks completed on the work described above and found to be satisfactory. All work accomplished I/A/W the Pilatus PC12/45 maintenance manual and is approved for return to service. END

Maintenance Release: This aircraft, airframe, engine, propeller or appliance was repaired and inspected in accordance with current regulations of the Federal Aviation Administration and is approved for return to service. All work was performed in accordance with the applicable aircraft, engine, propeller, and / or appliance maintenance manuals and instructions. Pertinent details are on file at this repair station under work order 16461.

Authorized Signature: Ben T. Games, Chief Inspector for FAA CRS# ECWR470D Sun Aviation, Inc. Municipal Airport, Vero Beach, FL 32961 (772) 562-9257 Fax: (772) 562-5396

Certifies that the work specified except as		WORK PERF, BY	CERTIFICATE OF RELEASE TO SERVICE Certifies that the work specified except as otherwise specified was carried out in accordance with JAR-14 and in respect to that work the aeroplane/aeroplane component considered ready for release to service.		
TOT. TIME		DATE WORK PERF.			
TOT. LDGS			JAR-145 Appvl. No: Authorisation No:		
COMPANY NAME			Company: Certifying Staff.		

Lions Air 4/98

Aircraft Log Book Entry Reg.: N466SA A/C hrs.: 247.3 PC-12/45 MSN 443 TSN 247:19 / TSO: N/A / LDG: 180 WO 816448 Date:21.07.03 to 06.08.03 **Work Report** 7398 Nose Landing Gear partially disassembled (on A/C); Tube and piston replaced; all seals replaced; NLG assembled and safetied. Tube (Stage2) pressurized with 57.5bar; Strut fitted with fluid. Mil 5606; Stage 1(low pressure) pressurized 3.5 bar. Leak check performed and findcheck accomplished. 1ea.bulb p/n: 972.72.59.502 from cabin light replaced. 1ea. bulb p/n: 972.73.19.109 from flood light replaced. Flood light cables new fixed. CACU battery p/n: 972..81.32.050 replaced. (next expiry date 07.2005) EIS p/n: 975.29.02.016 with s/n: 052 removed. (failure code 78) Exchange EIS p/n: 975.29.02.017 with s/n: 033 with TSN: 286.3 hr. installed. EFIS SG1 and SG2 configurated. acc. ECA-12-TM-02-032. Electrical cables installed acc. wiring diagram 843.44.12.002. Avionic system test (TCAS) AHRS No.2 fails intermitted; troubleshooting, cable loom and plugs from AHRS1&2 installation. Visual inspected as far as possible. Failure codes AHRS1&2 downloaded(no faults rec.) Volts and Amps at AHRS1&2 measured. o.k. AHRS1 and AHRS2 interchanged. Functional test carried out. JG-290 (EGPWS) new fixed SB, SL and AD: SL 068 Wind Vector/Speed and drift angle bug Display on the EFIS /EHSI SB 31-004Rev. A CAWS Modification to the Central Advisory and Warning Control System Relay K206 p/n: 974.20.04.211 replaced. (U/S) SB 28-011 Fuel Booster Pump Damage to wires during installation SB 79-004Rev. 1 Oil Pressure Transducer Replacement out Transducer p/n: 975.21.15.207 with s/n: 6563-6-21 in Transducer p/n: 975.21.15.214 with s/n: 6801-12-99; TSN: 0:00hrs. SB 27-015 Replacement of inboard and outboard flap flex shafts inboard rh; out p/n: 945.02.02.203 s/n: S0026 TSN: 247hrs.; inboard rh in p/n: 945.02.02.205 s/n: S5267 TSN: 0:00hrs. outboard rh out p/n: 945.02.02.204 s/n: S0013 TSN: 247.3 hrs. outboard rh in p/n: 945.02.02.206 s/n: S5254 TSN: 0:00 hrs inboard lh out p/n: 945.02.02.203 s/n: S0023 TSN: 247.3 hrs. inboard Ih in p/n: 945.02.02.205 s/n: S5264 TSN: 0:00 hrs outboard lh out p/n: 945.02.02.204 s/n: S0015 TSN: 247.3 hrs outboard lh in p/n: 945.02.02.206 s/n: S5269 TSN: 0:00 hrs I certify that this aircraft has been inspected in accordance as per AMM no. 02049 Rev.12 and was determined to be in airworthy condition. Refer to Work Report no7398 A&P no. 2651109 (B. Imboden) Certificate No./Type: Signature: Distribution: *Customer, UDA, UHF, UQ. *Please file this Document in your Aircraft technical log. Page 1 of

Aircraft Log Book Entry

Reg.: N466SA

PC-12/45 MSN 443

WO 816993 **Work Report** A/C hrs.: 275:12

TSN 275:12 / TSO: N/A / LDG: 196

Date: 03.11.03 to 07.11.03

UDA 7521

Completed a Annual inspection as per Pilatus Maintenance Manual document no. 02049 Rev.13

Schedule Maintenance: 2ea. Main Battery Service / emergency power supply capacity test performed./ flap actuators lubricated . i.a.w. MM ref. 27-50-00 / Flap actuators backlash check i.a.w. ref. 27-50-03 / Altimeters removed and tested / Static System Leak check / Transponder System check / Magnetic Compass check swing / LDG Bearings lubricated / Main LDG Bushes and Axles inspected / Pressurization outflow valve and safety valve removed and cleaned i.a.w. ref. 21-30-00 / Pressurization system test i.a.w. ref. 21-30-00 / Chip detector inspected / Servo mount removed and clutch test carried out i.a.w. ref. 22-10-00 / Power Plant Torque Limiter Test performed / Spark Igniters cleaned / Applied Airfoil de-icers surface coating / Fire extinguisher contents check / Functional Test Horizontal Stabilizer Runaway Aural Warning / Oil Filter Element P/N 968.20.12.588 replaced /

Additional Work: Rubber P/N 557.24.12.207 on R/H Main LDG Compartment replaced (defect) / 1ea. O2 Passenger Oxygen Mask P/N 535.21.12.141 replaced. (u/s) / Temperature Controller P/N 975.44.21.417 replaced. Out: S/N U0840; In: S/N U0628 (L/H AOA Plate Heater u/s) / 2ea. Power Supply P/N 972.72.59.503 replaced (u/s) Out: 27221, 27223; In: 33873, 15378 / 200 ltr. Fuel Jet A1 refilled.

SB's carried out:

SB 24-021 SB 24-022 SB 30-008 SB 79-005 R1 Electrical Power DC Generation / Gen 1 Ground sensing lead Electrical Power DC Power Replacement of Generator Control Unit Ice and Rain Protection, Pitot Static and AOA De-Icing Current Sensor Accessory Gear Box Chip Detector Installation

Crew Seat Pan performed

I certify that this aircraft has been inspected in accordance as per AMM no. 02049 Rev.13 and was determined to be in airworthy condition. Refer to Work Report no 7521.

SL 072

A&P no. 2468693IA (Risi Orlando)

Signature:

Certificate No./Type:



AIRCRAFT MAINTENANCE CH-6371 STANS / SWITZERLAND

and Domarks		Sianatura	Certificate No.
Make/Model: PILATUS PC-12/45			Reg No.: N466SA
Serial No. <u>443</u>	٠١٠		Date: <u>12-26-2003</u>
Time: <u>307.9</u>	BANYAN		
	Log Book Entry		
) EXCHANGED LCR-92, P/N 1241) INSTALLED CURRENT KLN-90B	0-2011,S/N 1310 OFF, S/N 9104 ON DATA BASE CARTRAGE		
he appliance on this operation had ederal Aviation Administration and epair station #SD4R514M.	is been repaired or tested and inspected in act is approved for return to service. Pertinent	ccordance with current re t details of the service ar	egulations of the e on file at this
Vork Order: <u>66478</u>	Inspector: O Authorized Signature:		
Banyan Air Service 1635 NV	F.A.A. Certified Repair Station SD4R514 V 51st Place Ft. Lauderdale, FL 33309 Phone:		4.491.6958
		and a land in the	
		7477	
			The second second

SUN AVIATION, INC., Vero Beach, FL (VRB)

Date: 4/26/04

Aircraft Log N466SA

EST Cycles:

Hobbs: 362.9 ACTT: 362.9 Pilatus PC-12/45 SN: 443

Accomplished a 100 Hour Airframe, Engine and Propeller Inspection using the Pilatus PC12 100 & 200 Hour Inspection Report as a guide. C/W FAR 91.207, PARA C and D, ELT Shock Test, next due at the Annual Inspection. C/W Fire Extinguisher re-weigh and no defects found. Accomplished AD 2004-04-01, Inspected Engine Driven Fuel Pump Housing and Valve Housing/Covers for fuel leakage and or extruding gasket material, noted PN RG9570R1/SN B4257 installed and No Further Action Required. Verified AD 2004-06-05 N/A by Manufacturers SN and No Further Action Required. Accomplished 100 HR Horizontal Stabilizer Trim Runaway and no defects noted, at this time. Removed all Life Vests for 2 YR Certification. Accomplished 6 MO Airfoil De-Icer Re-Coating Inspection; LH OUTBD, LH INBD, RH OUTBD and RH INBD. Accomplished 6 MO Battery Emergency Test/EPS Battery and Ops checked good, at this time. Serviced Oxygen to 1850 PSI. Installed new Washer, Pilot's side Glare-shield Retaining Bolt, under D.V. Installed (1) ea new Tail Nav Light Bulb PN 307. Serviced Hydraulic Reservoir with 5606. Profiled Fairing, Lower Trailing Edge of wing skin, RH Wing, Center Support. Replaced Nose Gear Actuator, installed overhauled PN 960.30.01.1110VH/SN 0065, cleaned area, no leaks noted and ops checked ok. Removed all Passenger seat base trim and drawers, adjusted locking mechanism on all passenger seats, reinstalled trim & Drawers and ops checked good. Installed new Chafe Tape, as needed, Cabin Door Threshold Plate. Removed RH and LH Main Landing Gear Struts, resealed both assemblies IAW Pilatus Maintenance Manual, pressurized assemblies with Nitrogen and Fluid, ops checked ok, reinstalled assemblies, performed Gear Retraction Test and ops checked good. Inspected LH Main Landing Gear Axle bolt and Bushings, measured all and found to be within tolerance, greased Bushings and Axle and reinstalled. Sumped Fuel Tanks and cleaned Fuel Filter. Performed Compressor Desalination and Performance Recovery Wash referencing P&W PT6A-67B Maintenance Manual.

Accomplished a 200 HR Wear Check, Starter Generator Brushes and no defects noted at this time.

Accomplished a 200 HR P3 Filter Cleaning and installed new "O" rings.

Accomplished a 200 HR Function Test, Microswitches/Prop Feathering and no defects noted, at this time.

Accomplished a 200 HR Turbine Wash.

Accomplished a 400 HR Wear Check # 2 Generator Brushes, found Brusher have 50% life remaining.

Accomplished a 400 HR Fuel Nozzle/Fuel Manifold Adapter Function Inspection by removing, shipping to P&W, Orlando, FL for inspection and reinstalling. Installed (2) New Fuel Nozzle Transfer Tubes PN 3011155. Installed (1) ea New Prop Seal PN 3030988. installed New Felt Strip, New Gasket and "O" Ring, Torqued, Safetied, ran engine, no leaks noted and function check !AW Manufacturers Specifications. Applied conductive cement to all four Prop Blades, dressed and painted prop.

MAINTENANCE RELEASE: This aircraft, airframe, aircraft engine, propeller or appliance identified above was repaired and inspected in accordance with current regulations of the Federal Aviation Administration and is approved for return to service. Al work was performed in accordance with applicable aircraft, engine and or appliance maintenance manuals and instructions. Pertinent details are on file at the repair station office, WO # 18668. A.D.'s researched through Rev. 4/12/2004.

CRS # ECWR470D Signature Sun Aviation, Inc, Municipal Airport, Vero Beach, FL 32961 (772) 562-9257

A PART

Aircraft Log Book Entry Reg.: N466SA A/C hrs.: 423:03 PC-12/45 MSN 443 TSN 423:03 / TSO: N/A / LDG: 272 WO 817643 Date: 25.10.04 to 29.10.04 = Work Report **UDA 7955** Completed a annual inspection as per Pilatus maintenance manual document no. 02049 Rev.14 - Schedule Maintenance: 2ea. main battery service / Emergency power supply capacity test performed / Flap actuators lubricated i.a.w. MM ref. 27-50-00 / Flap actuators backlash check i.a.w. ref. 27-50-03 / AHRS check swing / LDG Bearings lubricated / Main LDG bushes and axles inspected / Chip detector inspected / Servo mount removed and clutch test carried out i.a.w. Ref. 22-10-00 / Power plant torque limiter test performed / Spark igniters cleaned / Applied airfoil de-icers surface coating / Fire extinguisher contents check / Functional test horizontal stabilizer runaway aural warning / Applied Age Master - No.1 / Tq transducer and oil pressure transducer connectors cleaned / Propeller lubricated / Oxygen cylinder P/N 957.12.12.211 replaced due to hydrostatic test; out S/N ALT280-1860 TSN 423:03; in S/N ALT280-866 TSO 0:00/ Passenger oxygen masks inspected / Trend monitoring system, download performed Additional Work: 12 inch lamp P/N 972.72.59.502, interior lightning replaced / 1ea. bulb P/N 972.73.19.109 CB- panel replaced / Placard stabilizer trim P/N 511.24.12.048 replaced / Pitch servo clutch P/N 200-0318-010 S/N n/a replaced (disc broken) / NLG wheel bearing 959.51.81.975 replaced (corrosion) / Generator 1 voltage adjusted / 5ea shur lock P/N 944.81.18.201 replaced (u/s) / Windshield seal repaired / AOA seal repaired / EIS display unit P/N 975.29.02.017 replaced. - out S/N 052 TSN 423:03; in S/N047 TSN 282:20 (tq value not correct) / Interior fairing baggage compartment, repaired (cracked) / Teflon tape pax door hinge cover, replaced / 1ea. Mag-Lite installed / Supplied 2ea. Life West's AD's checked: Fuel booster pump wiring Carried out, Aug 06/03 AD 2003-20-15 Garmin GTX 330/330D mode S XPNDR N/A component not installed AD 2004-10-15 AD 2004-12-11 Replacement of flap flex shafts Carried out, Aug 06/03 Service Bulletin's: LDG inspection & replacement of Components SB 32-016 Part A performed MLG Actuator replacement (FOCA AD 2004-330) N/A by Actuator S/N SB 32-017 Introduction of dual navigation lights SB 33-009 Carried out Oxygen pipes - change routing positions (FOCA AD 2004-182) SB 35-004 Carried out Introduction of guard for the engine condition lever SB 76-001 Carried out - I certify that this aircraft has been inspected in accordance as per AMM no. 02049 Rev.14 and was determined to be in airworthy condition. Refer to Work Report no 7955. AIRCRAFT MAINTENANCE A&P no. 2468693IA (Risi Orlando)

CH-6371 STANS / SWITZERLAND

Signature: Certificate No./Type:

	IDCDAET IOC	NACCCA	6-17-05
Land Cont	Aircraft Reg:	N466SA PILATUS	Work Order #:
	Manufacturer:	PC12/45	F5060
	Model: Serial Number:		Pa 1 of 1
Aviation	Hour Meter:	455.6	AFTT: 455.6
770-458-9851	Tach Time:	400.0	TSMOH:
//0-458-9651	Total Cycles/La	ndinas 296	The second second
INSPECTION ON INSTALLED EQUIP. CA INSPECTION, ALL SATISFACTORY. CM DASH NUMBER. CAM A D. 2003-07-10 (N VERIFIED P/C/W S.B. ON 12-19-02. CM EXT. WT CHECK. C/W 2 YR. CAWS CAG JUNE/07. C/W 500 HR. COLD AIR REOIL MATRIX CHECK. C/W 1 YR. AUTOPILO INSPECTIONS. C/W 90 DAY/1 YR. EME ACTUATORS LUBE. C/W 6 MONTH DEIC C/W 2000 HR./1 YR. M/L/G AXEL & BUS STRUT (ldt p/m. 500.10.12.500), SERVIC WITH REPAIRED ASSY TSN: 1323.4 HR GOOD. ALL OPERATIONAL AND LEAK The aircraft, airframe, engine, propeller of accordance with current regulations of the service. Pertirent details of the spair and Signed:	V1 YR ELT BATT. FAR: CU CLOCK BATT. REPL L, WATER SEPERATOR T SERVO CLUTCH CHEC REGENCY BATT INSPEC CE BOOTS TREATMENT SHING INSPECTION. CN ED HYD. FLILID & N.2 PF S. (978.73. 20.003)(s/n of CHECKS SATISFACTOF T appliance identified abo The Federal Aviation Adminition on file at this repair state Output Description: Output D	21.207 INSPECTI ACE (972.81.32.0 CLEANING, AND K. C/W 400 HR/ TION. C/W 150 H C. C/W 1 YR. WHI V RESEALING OF IESS. REPLACET F: 0057)(s/n on: 0 YY. INTERPLACET WE was repaired a stration and is app ion under Work C EDUARDO R. N	ON. C/W-1 YR. FIRE 50)—NEXT DUE: HEAT EXCHANGER 1 YR. #1 & #2 BATT. R./1 YR. FLAP EEL BEARINGS LUBE. F L/H M/L/G SHOCK D FLAP P.D.U. ASSY 011), OPS CHECK C-SA and inspected in proved for return to Order No. 506 U.

Carry Forward

DUNCAN AVIATION...

N466SA Aviation Sales Pilatus PC-12

July 11, 2005

Ground checked Autopilot system and duplicated A/P failure. Reseated a loose connector at Autopilot Computer, P902. Autopilot system ground tested normal.

I certify that only the above described maintenance was accomplished and inspected, in accordance with manufacturer's specifications and current Federal Aviation Administration regulations, is airworthy and is approved for return to service. Pertinent details of the work performed are on file at this Repair Station under Work Order PJTBA.

repected by:	for Dur	ncan Avionics; Englewood, CO 8011
rspected by: RS # JG7D076J	Total Time:	Total Landings:
Daranced prop assembly using ACES Cleaned RH main wheel halves and	Pro Balancer Analyzer Model 202	0. Final IPS .09

noted. Installed new tire Michelin air p/n 025-350-0 SN: 2164W00093 and o-ring PN:MS28775-268.

MAINTENANCE RELEASE
The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No. 61844 Date: July 21, 2005 Signed for Skytech, Inc., PO Box 37507, Rock Hill, SC 29732.

CRS #C41R727N



580 Airport Road Rock Hill, SC 29732 803-366-5108 CRS #C41R727N

July 22, 2005

S/N:442

 N466SA
 Airframe Entry

 W/O 61844
 Hobbs:486.0

Lav pump motor does not run. Removed lav pump motor P/N: 5250-A S/N: P2391 and replaced with lav pump motor P/N: 5250-A S/N: P3966. Power checked lav; no defects noted

MAINTENANCE RELEASE

The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No. 61844 Date: July 22, 2005 Signed for Skytech, Inc., PO Box 37507, Rock Hill, SC 29732.

Skytech, inc.
LOG BOOK ENTRY

580 Airport Road Rock Hill, SC 29732 803-366-5108 CRS #C41R727N

N466SA W/O 61844

Airframe Entry Hobbs:486.0 July 22, 2005 S/N:442

Autopilot fails in flight. Troubleshot to faulty pitch trim adapter. Removed adapter PN 065-00164-0100, SN 1472 and installed adapter PN 065-00164-0100, SN 1107. Autopilot functional and ops check good.

MAINTENANCE RELEASE

The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the redair are on file at/this agency under Work Order No. 61844 Date: July 22, 2005 Signed for Skytech, Inc., PO Box 37507, Rock Hill, SC 29732.

Recorder



580 Airport Road Rock Hill, SC 29732 803-366-5108 CRS #C41R727N

N466SA W/O 61845 July 26, 2005 S/N:442

Removed Stick Pusher Computer P/N: SPWG92-3 S/N: 111495 for troubleshooting. Reinstalled Stick Pusher Computer P/N: SPWG92-3 S/N: 111495. Calibrated Stick Pusher Computer and AOA probes I/A/W Pilatus Maintenance Manual 22-20.

Airframe Entry

Hobbs:486.0

The aircraft and/or component identified above was repaired and inspected in accordance with MAINTENANCE RELEASE current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No. 61845 Date: July 26, 2005 Signed for Skytech, Inc., PO Box 37507, Rock Hill, SC 29732.

LOG BOOK ENTRY

580 Airport Road Rock Hill, SC 29732 803-366-5108 CRS #C41R727N

N466SA W/O 61845

an

Airframe Entry Hobbs: 486.0 August 17, 2005 S/N: 443

Removed the following equipment:

Nav 1 / Com 1 system components

Bendix/King KX 165A Nav/Com Transceiver, P/N 069-01033-0201, S/N 10744.

Installed the following components:

Garmin GNS 530A VHF Comm Transceiver / VOR/ILS Receiver / GPS Receiver system

- GNS 530A GPS Receiver, P/N 011-00285-01, S/N 46350466.
- GA 56 GPS Antenna Kit, P/N 010-10040-01, S/N 59095101.
- Eaton switch, P/N 17214-7936 (GPS 1/GPS 2 Map source select switch input to KMD 850) The Garmin GNS 530A VHF Communication Transceiver / VOR/ILS Receiver / GPS Receiver system Flight Manual Supplement, P/N 190-00181-04, Rev. G, FAA Approved on 8/3/2005, inserted in Airplane Flight Manual.

See FAA Form 337, dated July 20, 2005 and FAA Form 337, dated August 17, 2005 (GA 56 Antenna) for additional details.

W/O 61845

Page 1 of 2



580 Airport Road Rock Hill, SC 29732 803-366-5108 CRS #C41R727N

Bendix/King KDR 510 VHF Digital Link Receiver

- KDR 510 VDL Receiver, P/N 064-01088-0101, S/N 2975.
- KAC 503 Flight Information System (FIS) Module, P/N 071-00168-0311, S/N 2984. FIS Antenna, P/N FIS10-751-1.

See FAA Form 337, dated July 20, 2005 for additional details.

Weight & Balance and Equipment List updated and placed in the aircraft records.

New Empty Weight: 6.475.64 lbs. New C.G.: 234.21 in. New Moment: 1,516,646.74 in. lbs.

Additional Items:

- Changed wiring for existing Bendix/King KLN 90B GPS to re-assign it from GPS 1 to GPS 2.
- 2. Moved existing Bendix/King KT 70 (#1) Transponder and KR 87 ADF to center pedestal to allow for the installation of the Garmin GNS 530A GPS.

MAINTENANCE RELEASE

The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No.61845 Date: August 17, 2005 Signed Tim Ktsman for Skytech, Inc., PO Box 37507, Rock Hill, SC 29732. O CRS #C41R727N

W/O 61845

Page 2 of 2



580 Airport Road Rock Hill, SC 29732 803-366-5108

N424PB N4465A W/O 61845

Airframe Entry Hobbs:496.9

August 19, 2005 S/N:443

- Installed FCU PN 8063-042-01; SN 13093666 after compliance with P&W SB 14371. Ref Woodward WO 91333, dated 08/11/05.
- Performed P&WC S.B.No. 14371 Replacement of the fuel control unit, per section 3. "Accomplishment instructions". SB dated Apr 15/2005. FCU R&R performed per the P&WC PT6A-67B EMM chapter 73-20-00 and the Pilatus PC-12 AMM chapter 73-20-01-04. Performed rigging procedure per the Pilatus PC-12 AMM chapter 76-10-00-05, Page Block 501. Performed ground run procedures per the Pilatus PC-12 AMM chapter 71-00-00-05, Page Block 501.

MAINTENANCE RELEASE

The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No. 61845 Date: August 19, 2005 Signed William 16. Kalion for Skytech, Inc., PO Box 37507, Rock Hill, SC 29732 CRS #C41R727N

Recorder



580 Airport Road Rock Hill, SC 29732 803-366-5108

N466SA Airframe Entry W/O 61901 Hobbs:527.9 October 13, 2005 S/N:443

Repaired damaged pin for 232 data input for the MFD and GPS systems. Power checked GPS #1 and #2 systems and monitored along with MFD map display. No defects noted.
 Programmed EFIS systems to display Wind Vector Arrow from #2 GPS. Connected ADC 429 wires to #1 GPS and programmed to display Wind Vector Arrow. Power checked #1 and #2 GPS systems; no defects noted.

GPS systems; no delects floted.
 Cleaned and reseated connectors including antenna connectors on radar altimeter processor.
 Cleaned and reseated antenna connectors on antennas. Power checked radar altimeter; no defects noted.

Posented KMD 850 in tray. Power

defects noted.
 Removed KMD 850 and inspected connections on back. Reseated KMD 850 in tray. Power checked radar using KMD 850; no defects noted.

Stytech, inc. LOG BOOK ENTRY

580 Airport Road Rock Hill, SC 29732 803-366-5108

MAINTENANCE RELEASE

Recorder



580 Airport Road Rock Hill, SC 29732 803-366-5108

N466SA W/O 61969 Airframe Entry Hobbs:527.9 October 17, 2005 S/N:443

Removed #2 altimeter P/N 5305P-P50, S/N 32441 and installed overhauled altimeter P/N 5035P-P50, S/N 30887. See Mid-Continent WO# WRK0065708 for details of repair. Performed pitot static leak checks and altitude corolation. Ops check good.

MAINTENANCE RELEASE

The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No. 61969 Date: October 17, 2005 Signed from Skytech, Inc., PO Box 37507, Rock Hill, SC 29732. CRS #C41R727N

Skytech, inc.
LOG BOOK ENTRY

580 Airport Road Rock Hill, SC 29732 803-366-5108

C-CC---- No

I certify that the altimeter and static system test required by 14CFR § 91.411 and transponder test, including data correspondence, required by § 91.413, have been performed and found to comply with 14CFR part 43, appendix E and F.

Altimeter #1 model Altimeter #2 model Transponder #1 model Transponder #2 model	KEA346 5035P-P50 KT-70	Serial # Serial # Serial #	4561 30887 30572
Encoder #1 model Encoder #2 model	KT-70 KEA346 KEA346	Serial # Serial # Serial #	6539 4561 30887

241

Registration No.
3

and Remarks

Signature Certificate No.



580 Airport Road Rock Hill, SC 29732 803-366-5108

January 16, 2006

N424PB W/O 61984 Airframe Entry Hobbs:575.0

W/O 61984 Hobbs:575.0 S/N:442

1. Removed glareshield and repaired left corner side of glareshield IAW Pilatus MM. Reinstalled

glareshield.

The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No. 61984 Date: January 16, 2006 Signed CRS #C41R727N for Skytech, Inc., PO Box 37507, Rock Hill, SC

	Aircraft		RAFT LOG Recorder	
Date	This Date	Total	Reading	Maintenance
ی	Skyleck	LOG BO	OOK ENTRY	580 Airport Road Rock Hill, SC 29732 803-366-5108
	424PB //O 61984		frame Entry bbs:575.0	December 15, 2005 S/N:44 3
			above was repaired and	
of Sig	rrent Federal Aviation F	at this agency un	der Work Order No. 619	urn to service. Pertinent details 984 Date: December 15, 2005 PO Box 37507, Rock Hill, SC
of Sig	rrent Federal Aviation F the repair are on file a gned (())	at this agency un	der Work Order No. 619	urn to service. Pertinent details 984 Date: December 15, 2005

Skyle	1-1
1-10	CAD, inc.
Contre	0,0,0
Ng	LOG BOOK ENTRY

NARRSA

MAINTENANCE RELEASE

Box 37507, Rock Hill, SC 29732.

Registration No.

Rock Hill, SC 29732 803-366-5108 CRS #C41R727N

January 25, 2006

580 Airport Road

		Tuniditio Endy	Juliani J 25, 2000	
W/O 61984		Hobbs: 575.7	S/N: 442	-
1.	Removed	ailerons, rudder and elevator control surfaces due to hail d		
	a.	Removed aileron (RH) PN: 557.05.12.012 SN: 349, (LH)	PN: 557.05.12.011 SN: 349, Installed	-
		new aileron (RH) PN: 557.05.12.014 SN: 592, (LH) PN:	557.05.12.011 SN: 471.	
	b.	Removed elevator PN: 555.20.12.036, SN: 346, installe 619.	d new elevator PN: 555.20.12.039, SN:	
	C.	Removed rudder PN: 555.40.12.038, SN: 350, installed 617A.	new rudder PN: 555.40.12.038, SN:	
	d.	Installed new bushings, PN: 941.14.11.215, 941.14.11.	172, 941,14,11,161, 941,14,11,161,	
		941.14.11.170 and 513.27.12.051.		

Balanced all controls IAW Pilatus MM 51-60-02, 51-60-03 and 51-60-01.
 Installed controls IAW Pilatus MM 55-20-01-04, 55-40-01-04 and 57-60-01-04.

CRS #C41R727N

Airframe Entry

Installed four new static dischargers PN: 984.78.81.408.



580 Airport Road Rock Hill, SC 29732 803-366-5108 CRS #C41R727N

February 21, 2006

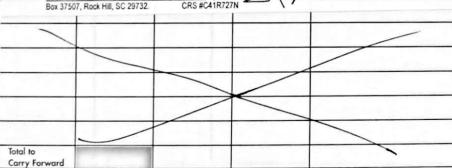
S/N: 443

N466SA Airframe Entry W/O 62151 Hobbs: 618.9

 Battery #1 amp indication fluctuates - Removed battery #1 current sensor PN: 524.52.12.720 SN: 302, replaced with new sensor same PN, SN: 373 IAW MM Ch. 24-30-08-04. Performed runup, NDN.

MAINTENANCE RELEASE

The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No. 62151 Date: February 21, 2006 Signed For 375/07 Rock Hill SC 29732 CRS #C41R727N





580 Airport Road Rock Hill, SC 29732 803-366-5108 CRS #C418727N

	6SA Air	frame Entry		March 2, 200
N/C	62151 Hobbs: 618.9	Total: 618.9.1		S/N: 443
	Performed a 100 hour inspection per the	Pilatus PC-12 A	MM chanter 05-20-10-0	n.
	Complied with ELT Inspection IAW FAR	91 207(d) Ratt	due 1/08	· ·
	Performed the following time limit items:			
	 A) - Secondary heaters therma 	I protection switch	hes testing /321,999	500 be)
	 vapor cycle cooling system 	n 15 min oneratir	on (2 weeks))	300 111)
	 Starter/Generator brushes 	insp. (200 hr)	(L HOULD))	
	 D) - Generator 1 & 2 ground po 	ints insp. (2 vrs)		
	 Horizontal Stab Trim Runa 	way Aural Wamir	na system functional tes	t (100 br. 1 ur)
	All foll devicers surface cos	sung. (b months)	g system randdonar (63	a (100 iii, i yi)
	 G) - Standby magnetic compas 	s swing (2 vr)		
	 H) - Propeller feathering micro 	switches (200 hr)	
	CW AD 2005-22-13 - Main landing ge	ar special bolt is	nenostica and a l	ment CW by malassass f
-	AUTOBIOL DILL	fliator replaces	nant CM 1/	1 (0)
	I/A/W service bulletin instructions. Ren	noved KSA 372	pitch actuator P/N: 065	DOES ES (OSE DO DO 404) ON
			prior, detadtor 1 /14. 003	-0030-30 (905.92.03.161) S/N:



580 Airport Road Rock Hill, SC 29732 803-366-5108 CRS #C41R727N

5160 and replaced with repaired KSA 372 pitch actuator P/N: 065-0056-59 (985.92.03.164) S/N: M4540 I/A/W Pilatus Maintenance Manual 22-10. Unit was repaired by Honeywell International Inc. on their W.O. A83772 dated February 9, 2006. Performed operational check of autopilot system I/A/W Pilatus Maintenance Manual 22-10. No defects noted. Removed Temporary Revision No. 11 from P.O.H

- 6. CW Pilatus SB 21-007 (Ref. 188) ECS by modification of the cable routing
- CW Pilatus SB 31-006 (Ref. 197) AHRS Removal of the jumper wire from connector J992. Performed aircraft electrical checks IAW S/B 31-006, NDN
- Troubleshot A/S indication to faulty diode terminal block TB 115-07. Removed Diode terminal block TB 115-07 and installed new diode terminal block P/N 971.42.31.880 at TB115-07 location. Performed system functional check IAW Pilatus MM 31-30 and POH. Ops check good.
- 9. Serviced Oxygen.
- 10. Adjusted #5 and #6 seat clutches.
- 11. Replaced both tail nav bulbs with P/N 972.87.87.184 (92-0802-0003). Ops check good.
- 12. Replaced belly beacon bulb with P/N 34-0226010-91. Ops checked good.
- Replaced belly deacon balls with P/N MS29513-232.
- Remove surface corrosion and treated in following areas: both MLG aft trunnion bolts, both MLG inboard attachments of MLG actuators, and both MLG actuator outboard rod ends...
- 15. Torqued terminal connections on K231 (Gen 2) relay.
- 16. Ad's checked through 2006-3.

I certify that this Aircraft has been inspected IAW a 100hr inspection and was determined to be in airworthy condition.

MAINTENANCE RELEASE The aircraft and/or component identified above was prairer and jospected in accumance with current Federal Aviation Regulations and was found airworthy for return to service. Destinent or air of the Mair are on file at this agency under Work Order No.62151 Date: March 2, 2006 Signed

Skytech, Inc., PO Box 37507, Rock Hill, SC 29732.



580 Airport Road Rock Hill, SC 29732 803-366-5108

N466SA W/O 62341		Airframe Entry Hobbs: 647.9 Total: 647.9 Cycles: 458	July 1, 2006	
	110 02041	11000S, 047.9 Total, 647.9 Cycles, 458	S/N: 443	
1	Performe	d annual inspection IAW PC-12/45 MM.		
2		spected ELT per FAR 91.207(d). Battery replacement due: Jan '08		
3	Performed	the following time limit items:		
	A)	- Cabin altitude switch functional test. (5 yr)		
	B)	- Cabin differential pressure switch functional test. (5 yr)		
	C)	Pressurization outflow & safety valve cleaning. (1500 hr, 2 yr)		
	D)	Pressurization system adjustment/test. (1500 hr, 2 yr)		
	E)	- Vapor cycle cooling system 15 min. operation. (2 weeks)		
	F)	- Servo mount clutch checks. (1000 hr, 1 yr)		
	G)	Ni-cad battery check and service. (1000 hr, 1 yr)		
	H)	- EPS capacity check. (1 yr)		
	n'			
	J)	- Horizontal Stab Trim Runaway Aural Warning system functional test (100 hr, 1 yr)		
		- Flap actuators lubrication. (w-400 hr, 1 yr) (b-150 hr, 1 yr)		
	K)	- Fuel low level warning system operation test. (1000 hr/4yr)		
	L)	- Air foil de-icers surface coating. (6 months)		
	M)	- Landing gear wheel bearing lubrication. (1 yr/tire change)		



580 Airport Road Rock Hill, SC 29732 803-366-5108

- Main gear axles and bushings insp. (2000 hr, 1 yr)

- Standby magnetic compass swing. (2 yr)

- AHRS swing, (2 yr)

Oil psi, engine torque and sitck pusher torque transducer connector cleaning. (500 hr, 1 yr)

Performed the following overhaul/replacement items:

A) - Fire ext. 1 yr contents check.

5. CW SB 53-004 (ref. 200) - Inspection of wing upper attachment lugs at FR21. Found drill mark damage on both sides of each wing lug measuring less than .3 mm deep each. Performed repair procedure #2 on all 4 damaged areas per SB instructions.

CW AD 2001-25-10 Effective 1/25/02 - Prevent current surges in the flap control system, Performed flap backlash measurements per AD para (d)(5). Next due 1247.9 ACTT.

A) RH OB: SN: 0649 .008 B) RH IB: SN: 0641 .008

LH IB: SN: 0639 .007 D) LH OB: SN: 0647 .008

Replaced LH main tire. Installed new Michelin 8.50-10 10ply tire SN: 5353W00377 IAW MM Ch. 32.

Re-glued loose insulation under glareshield.

Installed PM research protective boot PN: PM-190 on radome. Serviced oxygen IAW MM Ch. 35.

> -CRS # ECWR470D Sun Aviation, Ind. Municipal Airport, Vero Beach, FL 32961 (772) 562-9257

MERICAN WINTON

and | REG. N466SA MODEL PC-12/45

DATE

S/N 443 8/1/2006 HERNANDO COUNTY AIRPORT 2495 BROAD STREET BROOKSVILLE, FLORIDA 34604 PHONE (352) 796-5173 FAX (352) 799-4681

HOBBS 649.7 FAA REPAIR STATION #TZ4R562M

T.T.

W.O.

TACH N/A

6392

→ INSTALLED SPS-306 INVERTER UNDER CENTER FLOOR AREA F.S. INSTALLED FOUR OUTLETS, 3

IN CABIN UNDER TABLES, 1 IN COCKPIT SIDE ARM REST. 20 AMP CB INSTALLED IN NONESSENTIAL BUSS LABELED "115 VAC INV". OPERATIONAL CHECK OF INVERTER SYSTEM GOOD.

MAINTENANCE RELEASE The aircraft and/or component identified above way repaired and inspected in accordance with current instructions contained in Platus PC-12 the maintenance rules of the F.A.R.'s under which the operator is certified and is approved

for return to service as per these requirements. Pertinent details of the repair are on file at this agency under york order No. __6392 Auth. Mach. William Ofrens Auth Insp. William of the

Total to Carry Forward

and Re



580 Airport Road Rock Hill, SC 29732 803-366-5108

- 11. #1 battery failed capacitance ck. Deep cycled #1 battery and performed capacitance ck IAW Marathon MM Ch.
- 12. #2 battery failed capacitance ck. Deep cycled #2 battery and performed capacitance ck IAW Marathon MM Ch. 24, NDN.
- AD's checked thru 2006-12.

I certify that this Aircraft has been insp IAW an annual insp and was determined to be in airworthy condition.

MAINTENANCE RELEASE The aircraft and/or component identified above was repaired and inspected in accordance with current Federal Aviation Regulations and was found airworthy for return to service. Pertinent details of the repair are on file at this agency under Work Order No.62341 Date: July 1, 2006 Signed CRS #C41R727N Skytech, Inc., PO Box 37507, Rock Hill, SC 29732.

Date: 01/04/07

ACTT: 748.1 Hobbs: 748.1

Aircraft Log N466SA

SN: 443 Pilatus PC-12/45

Performed a 100 Hour Stab Trim Runaway Warning Aural Test IAW Pilatus Service Instruction 27-40-00 506 and

MAINTENANCE RELEASE: This aircraft, airframe, aircraft engine, propeller or appliance identified above was repaired and inspected in accordance with current regulations of the Federal Aviation Administration and is approved for return to service. All work was performed in accordance with applicable aircraft, engine and or appliance maintenance manuals and instructions. Perfinent details are on file at the repair station office, WO # 22479.

Signature

CRS # ECWR470D

Sun Aviation, Ind, Municipal Airport, Vero Beach, FL 32961 (772) 562-9257

AIRCRAFT LOG



Aircraft Reg. N466SA 5/11/2007 Manufacturer PILATUS Work Package: Model: PC12/45 T843 Serial Number 443 Pg. 1 o 2 Hour Meter 795.2 Total Cycles/Landing AFTT 795.2 605

CW 150 HR./ANNUAL INSPECTIONS PER PILATUS M.M. GUIDES. CAV CONTINUED AIRWORTHINESS INSPECTION ON INSTALLED EQUIP --- C/W PILATUS CAMPAIGN : PIL 2007-01* CARGO & PASS, DOORS STRIKER PLATE BOLTS TORQUE CHECKS, ALL SATISFACTORY, CW A.D. 2001-25-10 (JAN.25.02) 1000 HR/1300 FLT./1YR FLAP ACTUATORS BACKLASH CHECK, ALL SATISFACTORY, C/W A.D.2005-22-04 (DEC.12.05) PREVENT LANDING FAILURE--P/C/W PER S.B.32-016 ON 29-10-04 AT 423.3 HRS. C/W A.D.2006-16-13 (SEPT.18.07) FR21 WING LUGS INSPECTION----PICM PER S.B.53-004 ON 7-1-06 AT 647.9 HRS. C/W A.D.2007-03-06 (MAR.8,07) PREVENT PASS. SEAT LEGS FAILURE-CAV INSPECTIONS PER S.B.25-032, N/A BY SAVS. FOLLOWING A.D.'S LISTED N/A AD 2006-16-18, AD 2006-19-04, AD 2006-22-05,---- C/W 6000 HR/6YR. AIRFRAME CORROSION INSPECTION OF ALL AREAS, ALL SATISFACTORY, CAV 100 HR, 11YR, HORIZONTAL STAB TRIM WATNING TEST, CM 1YR, ELT ASSY FAR 91.207 INSPECTION-BATT DUE: 1/2008, C/W 2YR, CAWS BATTERY REPLACED (972.81.32.050), CW 3YR, PASS, O2 MASKS INSPECTION, CW 3 YR, O2 BOTTLE HYDRO-STATIC INSPECTION BY SAFETECH USA RS#UZYR394Y, OPS CHECK GOOD. CAN 5YR, OVERTEMP & DUCT OVERTEMP FUNCTIONAL TEST-- REPLACED DUCT SWT (973.81.32.212), OPSICHECK GOOD, CAV 2000 HR/5 YR FLT, CONTROL CABLES INSPECTIONS, CAV 2000 HR/1YR. MAJG AXES, & BESSHINGS INSPECTION, CAN 1YR WHEEL BEARINGS LUBE, CAN 150 HR/1YR, FLAP ACTUATORS LUBED, CW 1YR, EMERGRNCY BATTERY INSPECTION, CW 400 HR/1YR, BATTERY INSPECTIONS CF#1 5 #2. C/W 1000 HR/1YR. AUTOPILOT SERVO CLUTCHES INSPECTION. CONT.--.

Carry Forward

Registration No.

EPPS X

	Aircraft Reg:	N466SA	5/11/2007	
,	Manufacturer:	PILATUS	Work Package:	T843
	Model:	PC12/45	AFTT:	795.2
	Serial Number:	443	Tach Time:	
	Hour Motor:	705.2	Total Conflores	COE

CONT.----. C/w 6 MONTH DEICE BOOTS TREATMENT. REPLACED NOSE TIRE (021-327-0)(s/n off: 9310P00497)(s/n on: 5324W00107) AND 1ea. BEARING (13889) & 1ea. RACE (13830), REPACKED BEARINGS. REPLACED E.I.S. PANEL BATTERY (331KT-1), OPS CHECK GOOD. ALL OPERATIONAL AND LEAK CHECKS SATISFACTORY.

The aircraft, airframe, engine, propeller or appliance identified above was repaired and inspected in accordance with current requirements of the Federal Aviation Administration and is approved for return to service. Pertinent details of the Federal Aviation under work package no. TRA3

Signed:

EDUARDO R. VARGAS

for: Epps Air Service, Inc., ESM 688D, 1 Aviation Way, Atlanta, GA 30341

Pg 2 of 2

Maintananca	Recorder	Time		
Maintenance	Reading	Total	This Date	Date
			Brought Forward	
Date: 8-31-07			OBOOOK EHRLY	1.
	N 4665A		ilatus PC-12	_ P
enance manual P/N	bed in Pilatus mair	dures describ	omply with proce 2049 Pages 601 t	- b
enance manual P/N nance manual P/N etails are recorded	V Honeywell mair	hru 603, IAV Cev.8, Ground 1845	2049 Pages 601 t	0
nance manual P/N -	V Honeywell mair	hru 603, IAV Cev.8, Ground 1845	2049 Pages 601 t 06-15697-0008 F n S/O#	0. 0. 0. 0. F.

Carry Forward

		Registration No.				
	d Bangels			Signetury	Corp	cate No.
	AIRFRAME	September 13, 2007	Hobbs: 889.2	Total Time: 889.2	N466SA	
		Propeller Total Time: 8	89.2	Landings: 648		
		Air Conditioner: 229.1				
	Performed a	n Annual Inspection using	a Pilatus PC-12 che	ecklist and FAR 43, Ap	ppendix D.	
	Airworthiness D	Directive compliance check	ed through Revision	on 2007-18.		
	Complied w	ith 500 hour/annual engine	performance test.	No discrepancies not	ed.	
	Complied w	ith 500 hour/annual engine	torque limiter test	. No discrepancies no	ted.	
	Removed pro	opeller and sent for overha	aul. (See FAA For	m 8130-3 from Rocky	Mountain	
	Propellers, Inc.,	dated 09/06/07 placed in t	he aircraft records.	Reinstalled propelle	following	
	overhaul, Opera	ational and leak checks goo	od.			
	Complied wi	th 6000 hour/5 year pitch tr	im actuator overhai	al by removing Serial N	lo, 387 and	
	installing an ove	rhauled actuator, Serial No	o. 386. Total since	overhaul: 0.00. Time	since new:	
	1841.9 hours. D	ate of overhaul: 8/23/07. (See FAA Form 813	30-3 from Derco Repai	r Services,	
	Inc., dated 8/23/	07 placed in the aircraft re	cords.)			
	Downloaded	trend monitoring informa-	tion and forwarded	to The Trend Group.		
-	Removed an	d replaced ELT battery as	needed. New exp	iration date: 11/13. C	perational	
	check good.					
	Removed El	LT for inspection per FA	R 91.207 (d). R	einstalled ELT after	inspection.	
	Operational chec					
	Complied w	ith annual horizontal stab	ilizer trim runaway	aural warning opera	itional and	
	functional tests.	Operational and functions	al checks good.			
	Complied wi	th 15 minute run of vapor	cycle cooling syste	m. No discrepancies	noted.	
				I	Page 1 of 6	
_						

		AIRC	RAFT LOG		
Date	Aircraf		Recorder Reading	Maintenance	Registration No
AIRFI Co Co Co Re Mayo supply Pe Co inspec Re and re after c co install	mplied with 500 h mplied with 500 h moved emergency Aviation, Inc., date v. Operational chee rformed Generator mplied with 3000 tition per the manuf moved both aircraft assembled both bat charging and service mplied with 1000 ting an overhaulec	our cold air ECS our ECS water s our heat exchange power supply for d 8/29/07, placed ck good. #1 and #2 grou o hour #2 General acturer's specified batteries and per teries. Both battering. I hour starter/ger d unit, Serial N	onor Sturbine oil service. Reparator condenser cle ger inspection. r annual capacity checl d in the aircraft records. Indepoint inspection and recator drive bearing recations. Informed deep cycle. Deries capacity checked	N466SA caning. k. (See FAA Form 8130-3 from) Reinstalled emergency power	AIRFRAME (cont'd.) September 13, 2007 Complied with Pilatus A.D. 2001-25-10, by performing wing flap actuator backlash inspection in accordance with the manufacturer's maintenance manual with results as follows: Part Serial Previous Current Total Total Number Number Backlash Backlash Hrs Ldgs Rt Otbd 978.73.20.307 0649 Unknown .009 889.2 648 Rt Inbd 978.73.20.307 0641 Unknown .008 889.2 648 Lt Otbd 978.73.20.307 0647 Unknown .009 889.2 648 Lt Inbd 978.73.20.307 0639 Unknown .009 889.2 648 Aircraft total landings 648. Recurring inspection due each 600 hours, or annual inspection. Next inspection is due at 1489 hours, or 09/08. Complied with flap actuator lubrication service per Pilatus S.L. 080. Complied with 100 hour landing gear shock strut pressure check per Pilatus maintenance manual. Complied with 2000 hour/annual main landing gear axle and bushing inspection. No discrepancies noted. Complied with 2 year AHRS #land #2 calibration swing. Complied with 2 year standby magnetic compass calibration swing. Installed new compass correction card on compass. Complied with 2000 hour propeller feathering microswitch functional tests. No discrepancies noted. Complied with 500 hour/annual engine torque and stick pusher torque transducer electrical

AIRCRAFT LOG	
AIRFRAME (cont'd.) September 13, 2007 N466SA _ Complied with the annual fire extinguisher contents check. No discrepancies noted.	
Complied with Pilatus S.B. 24-023, 2nd Generator wiring modification, per the service - bulletin instructions. Complied with Pilatus S.B. 27-017, modification to the wiring circuit of the flap power drive	
unit, per the service bulletin instructions. Complied with Pilatus S.B. 34-013, airspeed indicator overspeed warning wiring modification, per service bulletin instructions.	
Complied with Pilatus S.B. 55-002, replacement of vertical stabilizer shear plate and strap, per service bulletin instructions.	
Adjusted aileron main cable tension per the manufacturer's specifications. Adjusted aileron autopilot cable tension per the manufacturer's specifications. Adjusted rudder main cable tension per the manufacturer's specifications.	

Complied with Pilatus S.B. 27-4 unit, per the service bulletin instru Complied with Pilatus S.B. modification, per service bulletin Complied with Pilatus S.B. 55per service bulletin instructions. Adjusted aileron main cable te Adjusted aileron autopilot cab Adjusted rudder main cable te Adjusted elevator main cable tension per the manufacturer's specifications. Replaced propeller spinner screw washers. Left and right card table mechanisms were cleaned, lubricated and adjusted per manufacturer's specifications. Operational check good. Serviced the lavatory as needed. Adjusted cabin seat #2 drawer per the manufacturer's specifications. Operational check good. Adjusted propeller low pitch warning switch per the manufacturer's specifications. Serviced air conditioner with refrigerant per the manufacturer's specifications. Leak check good

Carry

Removed air conditioner receiver dryer and replaced with new part. Leak check good. Page 4 of 6 ==== Registration No.

and Remarks Signature Certificate No. AIRFRAME (cont'd.) September 13, 2007 N466SA Removed wing ice detection light bulb and replaced with new part. Operational check good. Replaced three missing screws on left main wheel cover. Replaced missing right main wheel valve stem cover plug. Removed nose landing gear steering bracket and replaced with new part. Repaired the pilot and copilot windshield seal using Pilatus maintenance manual and PPG ____ Industries Service Information Letter Number SIL-172121-001. Removed left cabin window seal #3 and replaced with new part. Removed right cabin windows seals #1,#3,#4 and #5 and replaced with new parts. Removed cargo door window seal and replaced with new part. Removed left inboard flap fairing mount bracket cherry max rivets and replaced with new Removed nose wheel seals and replaced with new parts. Leak check good. Painted and installed new vertical stabilizer stinger flange fairings. Secured coaxial cable under interior side panel at cabin seat #3. Complied with Pilatus S.B. 32-020, main landing gear special bolt inspection and replacement. Not applicable by bolt serial number. 600 Complied with Pilatus S.B. 71-007, torque oil pressure pipe and hose assembly in the complete of the complete and replacement, by inspection per the service bulletin instructions. Page 5 of 6

AIRCRAFT LOG						Registration No
Date	Aircra This Date	ft Time Total	Recorder Reading	Maintenance	and Remarks	Certificate No
prope the m Co as req Dy the Pi	eller speed indication anufacturer's specification omplied with continuitied. ynamically balanced latus maintenance r	ell S.L. HC-SL-6) in to be accurate. fications. nucd airworthines d the propeller w manual. aft has been inspe	1-254, propeller operatin Engine rigging for idle as inspection requirement ith Chadwick/Helmuth of	N466SA ag restrictions, by verifying the speed was found to be within ats for the installed equipment equipment and procedures per an Annual Inspection and was		AIRFRAME October 12, 2007 Hobbs: 889.8 N466SA Removed generator #2 drive adapter assembly and replaced with new parts. Operational check good. For Aviation Sales, Inc. W.O. 15517 Glenn D. ones, A&P 3130913 AIRFRAME October 12, 2007 Hobbs: 890.3 N466SA Installed tow bar head holder with tow bar head per Aviation Sales Drawing PC 12030 and Form 8110-3 dated 6/25/96. (See FAA Form 337 dated 10/12/07. Amended weight and balance information and equipment list as required.
	viation Sales, Inc. 15477		Glenn D. Jones	A&P 3130913 IA Page 6 of 6		For Aviation Sales, Inc. Mac D. Kleinschrodt, A&P 476802692 W. O. 15540 Max D. Kleinschrodt, A&P 476802692
to Forward						

SO#	07-12486		DATE	11/14/07		
REG.	N466SA		Lincoln Municipal Airport [LNK]		ACTT	891.9
MAKE	Pilatus	MODEL	PC-12/45		HOUR	891.9
	1 84	SERIAL NO.	443		CYCLES	

- Complied with RVSM requirements IAW Honeywell STC #SAO253WI-D. Inspections and testing was completed IAW Honeywell Manual Number 006-20034-0000 and IAW Honeywell STC SA01253WI-D. See FAA form 337 dated 11/14/2007 for details.
- Modified left outer pilots canted crossbar panel to accept standby altimeter P/N. 5035P, S/N. 36887. Modified copilots panel to accept copilots AM-250 altimeter P/N. 70256N02D02, S/N. 07090850D. Installed pilots AM-250 altimeter P/N. 70256N02D02, S/N. 07090849D. Installed A/P-ALT source select switch with 1 each minor deviation to the STC, installed Annunciator switch into existing unused Annunciator position in copilots panel and placarded as required. IAW Honeywell dual AM-250 STC #SA01116WI-D. Installed relay bracket installation IAW Honeywell dual AM-250 STC #SA01116WI-D Electrical installation was installed IAW Honeywell Dwg. 159-08290-2052. Mechanical Installation completed IAW Honeywell Dwg. 159-08290-5100. These drawings included in the Honeywell dual AM-250 STC #SA01116WI-D. See FAA form 337 dated 11/14/2007 for details.
- 3. Removed existing KDR-510 P/N. 064-01088-0101, S/N. 2975 and modified mounting tray, using 2 ea. MS21059L08 nut plates installed with CCR264SS3-3 rivets, in order to install the new KDR-610 radio. Removed old KDR-510 antenna, P/N: FIS10-751, bagged and stowed its connector. Fabricated a seal patch at old antenna hole using 063-2024-T3 alclad aluminum as a minor alteration. Fabricated and installed one antenna doubler out of .040 2024-T3 alclad aluminum IAW Silverhawk Aviation drawing number SHSPC12-0001 Rev I/R see FAA form 8110-3 dated 11/13/2007 for details. Installed the Comant XM satellite antenna, P/N. CI 420-1, S/N. 172288, IAW antenna manufacture instructions. Installed Honeywell KDR-610 P/N. 4092743-902, S/N. 32371358 and associated wiring IAW Honeywell KDR-610 receiver interconnect drawing No. 155-01811-0000. Function test completed IAW Bendix/King KMD 550/850 Manual 006-18237-0000 Rev. 5 Dated March, 2007. See FAA form 337 dated 11/14/2007 for details of the Cormant XM satellite antenna doubler installation.
- Fabricated an installation mounting tray out of .050 2024-T3 alclad aluminum and installed it under the center walkway or an existing unused avionics mount position, 3 inches. aft of frame station 20, on the centerline of the aircraft_using CCR264SS3-3 and CR3213-4-3 rivets. Installed new Aircell ST-3100 communications system unit P/N. 400-10680-002 S/N. 1068021168 IAW Aircell ST3100 Mechanical/ Antenna Installation Manual No. 800-10355. Installed Aircell Handset P/N. P12198-001, S/N. None and Cradle P/N. 400-10038-001, S/N. None IAW the Aircell Handset Installation Guide to, P/N: 800-10055 Rev. D, to the co-pilots cockpit non-structure divider forward upper left hand corner using four NAS1836-D8-11 inserts with DP420 adhesive and supplied hardware. Re-utilized the existing GPS Iridium

Registration No.

SO#	07-12486		SILVERHAWK A	VIATION INC.	DATE	11/14/07
REG.	N466SA		Lincoln Municipa	I Airport [LNK]	ACTT	891.9
MAKE	Pilatus	MODEL	PC-12/45		HOUR	891.9
	7.07	SERIAL NO.	443		CYCLES	

antenna, P/N: S67-1575-109 S/N: 859 and existing coax after verifying compatibility with the Aircell ST3100 system. Installed Aircell ST3100 system IAW Aircell electrical Dwg 655-10680. See FAA form 337 Field Approval dated 11/14/2007.

- Removed existing GNS 530 unit P/N. 011-00940-00, S/N. 46350466 and sent to Garmin for WAAS upgrade. Installed modified GNS 530W P/N. 011-01066-40, S/N. 46350466. See FAA form 8130-3 dated 10/31/2007 for details. Installed new GA35 antenna IAW manufacturers instructions. Completed GNS -530W installation IAW Garmin GNS-530W upgrade Manual P/N. 190-00357-06 Rev (A) and STC No. SA01933LA. See FAA form 337 dated 11/14/2007 for details.
- Reset default code to 1200 IAW Bendix King KT70 installation/maintenance manual No. 006-00673-0002.
- Installed provisions for cabin microphone and headphone jacks into non-structural interior card tables outer/Velcro shells forward left and forward right card tables as a minor alteration to a non-structural interior panel IAW AC-43.13-18 Sec. 3 plastics pg 3-17 and 3-18. Installed and wired in microphone and headphone jacks P/N. SWCS12B and P/N. SWC 12.3. Installed audio input jack P/N. 27-655 into right hand card table above aft Mic/ phone jacks. Jacks wired into existing audio comm Panel GMA340 IAW GMA 340 Manual No. 190-00149-01.
- Chip between left hand static port mount screw. Cleaned and painted area as required, around left static port to comply with RVSM inspection/ static port requirements.

The work identified above was repaired and inspected in accordance with current Federal Aviation Regulations and is approved for return to service as airworthy.

Pertinent details of the repair are on file at this agency under work order No. 07-12486 Date November 16, 2007

Signed Scotty G. Long
Silverhawk Aviation Inc. CRS# YQSR759U
Lincoln, NE 68524

	Aircraft Time		Recorder	
Date	This Date	Total	Reading	Maintenance

1466SA			AVIATION INC.	DATE	11/14/07
Pilatus	MODEL	DC 12/45	al Airport [LNK]	HOUR	891.9
					Prince Co.
-			ilatus MODEL PC 12/45	ilatus MODEL PC 12/45	466SA Lincoln Municipal Airport [LNK] HOUR PC 12/45

I certify that the Altimeter, Static and ATC Transponder system tests required by FAR 91.411 and 91.413 were performed
and found to meet the accuracy requirements of FAR Part 43 appendix E and F. The Altimeter was functionally tested to
35,000 feet IAW Honeywell Group RVSM approval manual 006-20034-0000 sect. 3. Results were recorded on Table 3-3,
on page 3-11.

#1 Encoding Altimeter: #2 Encoding Altimeter: Stand-By Altimeter:	P/N 70256N02D02 70256N02D02 5035P-P	S/N 07090849D 07090850D 36887
#1 Transponder:	KT-70	30360
#2 Transponder:	KT-70	30572

The work identified above was repaired and inspected in accordance with current Federal Aviation Regulations and is approved for return to service as airworthy Perlinent details of the repair are on file at this agency under work order No. 07-12486 Date 11-14-07

Signed Silverhawk Anation Inc. CR\$# YQ\$R759I

Total t	0
Carry	Forward

Registration No.____

	00.40707	7.1	SII VERHAWK	AVIATION INC.	DATE	02/04/08
SO# REG.	08-12707 N466SA			al Airport [LNK]	TACH	939.0
MAKE	Pilatus	MODEL	PC-12/45			
		SERIAL NO.	443			

Install Flight Display Systems cabin entertainment system.

Removed existing Sony FM/AM/CD player Model No, P/N. CDX-CA660X, S/N.3512599.

Installed Flight Display Systems DVD/CD/MP3 player Model No, P/N. FD932DVD-4B, S/N. 7716725 into the existing
mount assembly in the right hand cabin entertainment pyramid cabinet. New DVD player mounted with the included
hardware.

Installed Moving Map Computer P/N. FD200CPU, S/N. 7007112 to the forward side of the right hand cabin entertainment pyramid cabinet by potting four (4) each inserts P/N. NAS1836-08-07 with DP-420 adhesive. Moving Map Computer installed with four (4) each screw P/N. AN525-832R7 and four (4) each washers P/N. MS35338-43 into the potted inserts.

Installed Flight Display Systems (carry on, to be stowed during take-off and landings) display/monitor P/N. FD102ARM-FM, S/N. 7314010B, mount P/N. FD BASE-W, S/N. None to the right hand cabin non-structural side wall at FS260.5, just aft of the Emergency Exit door 5 inches above the right hand drink rail. Display/Monitor Mount installed with a .040 2024-T3 aluminum doubler secured with Hysol adhesive P/N. U000701, the mount was installed with the supplied

hardware.

6. Fabricated and installed circuit breaker mount/angle for the new cabin entertainment system bussing. Mount/angle fabricated out of .040 2024-T3 aluminum, prepped and epoxy primed. Installed two (2) each potted inserts P/N. NAS1836-06-07 with DP-420 into the existing entertainment shelf pallet assembly in the right hand cabin entertainment pyramid cabinet DVD/CD/MP3 player shelf. The mount/angle was installed with two (2) each screws P/N. AN526-632, into the potted inserts in the shelf. A hole for access to the circuit breakers was provided in the bottom of the shelf and edge

 Installed Flight Display Systems Camera P/N. FD600CAM-2, S/N. 7701050 to the non-structural flight compartment co-pilot side instrument panel glare shield 6 inches right of aircraft center line with four (4) each screws P/N. MS35214-43, four (4) each washers P/N. AN960-8L and four (4) nuts P/N. MS21044N08.

Installed wiring, coax and circuit breaker protection per Silverhawk Aviation Drawing No. SHA-PC12-002 Rev. P1see FAA form 8110-3 dated February 4, 2008 and FAA form 337 dated February 4, 2008 for details.

50#	08-12/07		SILVERHAWA A	VIATION INC.	DATE	02/04/08
REG.	N466SA		Lincoln Municipa	Airport [LNK]	TACH	939.0
MAKE	Pilatus	MODEL	PC-12/45		43-4	
	1 1	SERIAL NO.	443	(Marial)		

CH LEDUAUN ALMATION INC

Complied with Pilatus Service Bulletin Number: 23-004 dated May 08, 2003 by incorporating Modification Kit Number 500 50 12 280

10. N1 and N2 annunciator bulbs inoperative. Installed new light bulbs P/N, 6839, lights function check good.

11. Terrain fail on Garmin GNS 530W. Installed correct terrain/TAWS data card system checks good on ground power

Performed Garmin GNS 530W data base update cycle 0801 expiration date February 14, 2008

......

Reconfigured RS232 port 1 to proper data configuration.

14. ILS inoperative on EFIS display, reconfigured signal Generator- ILS needles appear on EFIS display.

The work identified above was repaired and inspected in accordance with current Federal Aviation Regulations and is approved for return to service as airworthy. Pertinent details of the repair are on file at this agency under work order No. 08-12707 Date February 4, 2008

Silverhawk Aviation Inc. CRS# YQSR

Signed Scotty G. Long

Lincoln, NE 68524

SO#	07-12626		SII VERUANIE ALE			
REG.	N466SA		SILVERHAWK AVIATION INC.		DATE	12-16-07
MAKE	Pilatus	MODEL	Lincoln Municipal Airport [LNK]		HOUR	913.1
	1 matus		PC 12	ACTT:	TACH	010.1
46-24-7		SERIAL NO.	443		CYCLES	N/A

Complied with Garmin Software Service Bulletin No. 0740 Rev. A on GNS-530W S/N: 46350466. This upgrades the software to level 3.0. System function check was completed in accordance with service bulletin instructions.

The work identified above was repaired and inspected in accordance with current Federal Aviation Regulations and is approved or return to service as airworthy. Pertinent details of the repair are on file at this agency under work order No. 07-12626 Date December 16, 2007

> Silverhawk Aviation Inc. CRS# YQSR759U Lincoln NF 68524

DATE 02-04-08 SILVERHAWK AVIATION INC. 08-12743 SO# Lincoln Municipal Airport [LNK] TACH 939.0 REG N466SA MODEL PC12/45 MAKE Pilatus SERIAL NO. 433

Desired No. 2001 SEASONS A.

Radar Inoperative passes test but no display in-flight. Removed radar ant/rcvr/xmtr and sent to Bendix King for repair. Received radar antenna in a modified status. Unit checks good during test mode on ground power.

The work identified above was repaired and inspected in accordance with current Federal Aviation Regulations and is approved for return to service as airworthy. Pertinent details of the repair are on file at this agency under work order No. 08-12743 Date 02-04-08

Silverhawk Aviation Inc. CRS# VOSP: Lincoln NE 68524

AIRFRAME Removed flan	February 21, 2008	Hobbs: 962.7	N466SA
_ manufacturer's spe	actuators for freeze test. Flap actual reaker 955 and K249 relay with no crifications. Operational checks go	ew parts. Rigged and adjusted	l flaps per the
Shimmed DV	window latch per the manufacture	r's specifications. Leak chec	k good.

For Aviation Sales, Inc.	
W O 15671	

Registration No
The second secon
and Remarks
AIRFRAME (cont'd) March 21, 2008 Complied with A.D. 2007-21-11 and Pilatus S.B. 71- inspection and replacement. No further action required Pilatus A.D. 2008-02-03 has been previously compl S.B. 32-020, main landing gear bolt inspection. Complied with Pilatus S.L. 111 and Pratt & Whitn service bulletin instructions. Not applicable by inspecti Removed, cleaned, resealed and reinstalled the right the manufacturer's specifications. Operational and leak Bled the left brake system per the manufacturer's sp Removed propeller low pitch warning K051 relay a good. Tightened oxygen bottle service fittings per the man Adjusted passenger cabin seat #1 recline per the m good. Adjusted passenger cabin seat #4 swivel per the m good.

Carry Forward

Signature N466SA 1-007, torque oil pressure pipe and hose assembly plied with on 09/13/07 at 889.2 hours per Pilatus tney S.B. 14410, spark igniter inspection, per the ction. nt brake assembly. Bled the right brake system per ak checks good. specifications. Operational and leak check good. and replaced with new part. Operational check anufacturer's specifications. Leak check good. manufacturer's specifications. Operational check manufacturer's specifications. Operational check Page 2 of 3

Certificate No.

	Aircraft	Time	Recorder	
Date	This Date	Total	Reading	Maintenance
	Brought Forward			
IDEDA	ME (cont'd) Marc	h 21 2008		N466SA
Service Service Compared ind	checks good. ced main gear struts w ood. olied with Hartzell S.L lication to be accurate	rith hydraulic fl HC-SL-61-25 e. Engine riggi	uid and nitrogen per the 64, propeller operating re ing for idle speed was fo	Goodrich instructions. Operational manufacturer's specifications. Leak estrictions, by verifying the propeller bund to be within the manufacturer's
I cert	ify that this aircraft ed to be in airworthy	has been inspected condition.	ected in accordance wi	th a 100 Hour Inspection and was
determin			-1///6/	2337/20

W. O. 15693

Total to Carry Forward

and Remarks				book In	Signature		Certificate No	
				ANIATION INC		DATE	04/10/08	
O#	08-12926		SILVERHAWK	AVIATION INC.		ACTT		
of the second				pal Airport [LNK]	443	Tach	979.0	
REG.	Pilatus	MODEL	PC-12/45	SERIAL NO.				
2. Rep	programmed GN and pin G at TB2	S530 to reflect Pi 70-04 not mated ht display system	with new welcome	ected pin into terminal l logo. Also reformatted	map font sizi	ing. Checks g		
5. Au cor	ver. dio levels from D nfortable to lister	VD/CD player wit	thin norms adjusted	d volume at front of DV previous install- remov	D/CD player, ved capped ar	max setting o	arely used wire-	

Registration No.____

audio panel back light checks good.

Replaced burnt out light bulb. GPS 1 illuminates on map switch. P/N off: LTX 3900/0. P/N on: LTX 3900/0.

Re-secured Velcro tabs attaching card table face cards to side wall.

Installed IR DVD remote wiring. Checks good on ground.

The work identified above was repaired and inspected in accordance with current Federal Aviation Regulations and is approved for return to service as airworthy. Pertinent details of the repair are on file at this agency under work order No. 08-12926 Date 04/10/08 Date 04/10/08

- PINGIN NICE	Pilatus Aircraft Ltd. March 21, 200	Registration No	
DUNCAN N466SA Hoiles, Timothy	Pilatus Aircraft Ltd. March 21, 20' 443	and Remarks	Signature Certificate No.
Removed defective Garmin GMA 340 Audio P 011-00401-10, s/n: 96264855). Ground test op	ancl ($p/n;011\text{-}00401\text{-}10,s/n;96264505$) and installed rental GMA340 (p reration demonstrates good.	n:	
I certify that only the above, described mainte specifications and current Federal Aviation At Pertinent details of the work performed are on f Inspected by: CRS # JG 701761	enance was accomplished and inspected, in accordance with manufacture dministration regulations, is airworthy and is approved for return to servi ile at this Repair Station under Work Order VSRQA. for Duncan Avionics; Englewood, CO 80112 Total Time: Total Landings:	Systems, Inc., dated 02/20/08 placed in drive belt and replaced with new part. C	compressor condenser motor Serial No. 2534 No. 495. (See FAA Form 8130-3 from Enviro the aircraft records.) Removed compressor
DUNCAN N466SA AVIATION Aviation Sales	Pilatus PC-12 April 18, 2008		
emoved the loaner audio panel, P/N:011-0 paired/serviceable audio panel, P/N:011-0 ecifications on the ground.	1-00401-10, S/N:96264855 and reinstalled the customer's original 401-10, S/N:96264505. The audio panel operationally checked to	AIRFRAME June 9, 2008 Removed overhead panel Serial No. JWPIL147 JWPIL336. (See EASA Form 1 from Pilatus Faircraft records.) Operational check good.	
eccifications and current Federal Aviation Adv	ance was accomplished and inspected, in accordance with manufacturer ministration regulations, is airworthy and is approved for return to service le at this Repair Station under Work Order VSRQA.	Tor Aviation Sales, Inc.	Onald C. Gish, A&P 523046946
rspected by: C	for Duncan Avionics; Englewood, CO 80112 Total Time:Total Landings:	The second of professional participation of position	

	1		RAFT LOG	1000	Registration No					
Date	Aircraft This Date	Total	Recorder Reading	Maintenance	and Remarks	Signature	Certificate No			
	Brought -				-1		My Call I			
-28-2009		work or	er 1892 - Cont	land Acad let	This aircraft Las been Inspected as	1	4. 1.1.1			
18,2008	For details	/) 1	1 , 1 ;	, way with Acro Mat	COS-BOL in accordance with	11/1111111111111111111111111111111111	Late / October 14			
	also in en	try way o.	cockeit. Aut	75667 Ry quel	A BUSINESS OF THE STATE OF THE	Willia Shur	AP387589360			
	cockpit.	/	// // //	2966Rel on cabin,		0				
	service H	, date	Willia Sil	1 A397549366						
	AIRFRAME J	une 26, 2008	Hobbs: 102	7.3 N466SA - ts. Operational check good.						
	Leak check good.	ut with hydrauli	ic fluid and nitrogen per plug with new part.	ts. Operational check good. the manufacturer's specifications.						
to ry Forware	For Aviation Sales, W.O. 15768		MEB.	AP235792						
	13/00		Mark E. Biron, A&	P 2387920			Y 100 100 100 100 100 100 100 100 100 10			

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4300 S KENNEDY ST., BOISE, IDAHO 83705-8132, U.S.A

PHONE 208-338-1800

ATE: 29 Dec 2008	AIRCKAF	I LUGBOOK IN	SFRT
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UKCKAFI MAKE	PII ATTIC	MODEL	

- MINCHELL I MINKE	P	LAIUS		MODEL	DC.	-12/45	
N NUMBER	N466SA	S/N	443				
Completed Annual Airforn		3/14	443	HOBBS:	1086.1	LDGS:	740 .

mpleted Annual Airframe and Engine inspections using Western Aircraft checklist Form MX-007 Rev 10 as per Pilatus PC-12 IETP Rev 10 dated 1 March 2008, document number 02049 Issue 1, Rev 0, dated 20 Nov 2007, section 12-A-05-20-20-00A-281A-A, and Pratt and Whitney Canada Model PT6A-67B Maintenance Manual, part number 3038336, Rev. 33, dated 7 Nov 2008, section 72-00-00, table 602(minor, routine, and 100 hour inspection items). Accomplished the requirements of AMM 12-A-04-00-00-00A-

Chapter 4 & 5 Items addressed at this time

-

180 day airfuil deicer surface treatment is accomplished by application of Age Master No. 1 per AMM12-A-05-10-20-00A-281A-A requirement.

12 month fire extinguisher inspection and contents check is accomplished per AMM12-A-05-10-10-00A-281A-A requirement.

12 month ELT inspection (S1821502-02)is accomplished per FAR 91-207 (d) (1) (2) (3) & (4). No discrepancy noted. Battery change is due 11/2013.

12 month emergency power supply capacity check PS855B S/N 3131 is accomplished per L-3 SL 170. Capacity is greater than 100%. Reinstalled and tested per AMM 24-52-

12 month landing gear wheel bearing lubrication is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-12-20-06-00A-902A-A.

12 month WAI recommended landing gear shock strut hydraulic and nitrogen service check is accomplished per AMM 12-A-32-10-00-00A-902A-A and AMM 12-A-32-20-00-

12 month/200 Hour spark igniter inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement, AMM 12-A-74-20-00-00A-920A-A and EMM 74-20-00

12 month/400 hour fwd battery (P/N 976.31.302) service is performed per AMM12-A-05-10-20-00A-281A-A requirement and Marathon CMM BA 24-34-00 Rev 3. The hattery S/N 10103557 is deep cycled at this time. Installed (3) new manufacture cells P/N 30114-01C in positions 6, 11, & 15.

12 month/ 400 hour aft battery (P/N 976.31.302) service is performed per AMM12-A-05-10-20-00A-281A-A requirement and Marathon CMM BA 24-34-00 Rev 3. The

12 month/400 Hour propeller lubrication is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-12-20-05-00A-902A-A.

12 month 500 Hour torque limiter check is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM12-A-71-00-00-00A-903A-A, Table 501)

12 month/500 Hour torque pressure and stick pusher torque transducer electrical connector cleaning is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement

12 month/ 600 Hour chip detector inspection is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and EMM 72-00-00.

12 month /2000 Hour main landing gear axles and bushings inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM12-A-22-10-00-00A-313A-A

12 month/3000 hour horizontal stabilizer trim aural warning function test is accomplished per AMM12-A-04-00-00-00A-000A-A

12 month/3000 hour horizontal stabilizer trim aural warning function test is accomplished per AMM12-A-04-00-00-000A-000A-A

1 year RVSM conformity inspection is accomplished per CMM 006-26034-0000Rev C Para 5.3.1. Corrected paint transitions in the RVSM critical area to conform the RVSM conformation in the RVSM critical area to conform the RVSM

2 year Generator 81 and 82 grounding point inspections are accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-24-30-00-00A-901A-A.

2 year CAWS CACU clock battery replacement is accomplished per 12-A-05-10-10-00A-281A-A requirement and AMM 12-A-31-50-03-00A-920A-A 2 year AHRS system calibration & stand-by compass checks are accomplished per 12-A-05-10-10-00A-281A-A requirement and AMM 12-A-31-30-03-00A-280A-A. (continued on the next page)

AIRCRAFT LOGBOOK INSERT(continued)

PC-12/45 MODEL

Pg 2 of 3

AIRCRAFT MAKE HOBBS: 1086.1 LDGS: 740 S/N 443 N466SA 2 year/1500 Hour pressurization outflow valve and safety valve cleaning is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-30-01-00A-

2 year/1500 Hour pressurization system adjustment and test is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-30-00-00A-903A-A.

Replaced VCCS condensation drain grommet and removed obstruction from fwd fuselage belly drain as required. Performed cabin leak rate test. AMM 21-30-00-00A-903A-A. Leak rate is satisfactory.

5 year cabin altitude switch function test is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-30-07-00A-920A-A. Reference

details below for each cabin switch. 5 year cabin pressure differential switch test is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-30-08-00A-903A-A.

200 Hour P3 air filter cleaning is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and EMM 73-10-07.

200 Hour propeller feathering switches tests are accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-76-10-02-00A-903A-A.

400 Hour fuel manifold adapter, nozzles and boroscope inspections are accomplished per AMM12-A-05-10-20-00A-281A-A requirement and EMM 73-11-05. Performed satisfactory operation and leak checks per AMM 12-A-71-00-00-00A-903A-A. Reference engine log entry for details.

1000 Hour fuel cross vent and outward check valve operational test is accomplished per AMM 12-A-28-10-01-00A-903A-A.

2000 Hour first stage compressor inspection is accomplished per EMM 72-30-05 page block 201. Reference engine log entry for details. 2000 Hour hot section inspection is accomplished per AMM 12-A-05-10-20-00A-28 A-A requirement and EMM 72-00-00-06 table 604. Reference engine log entry for details.

Airworthiness Directives and Service Bulletins addressed at this time

AD 2008-07-11 Amdt 39-15452 dated 9 May 2008, complied with by the insertion of specified documents into the FAA approved maintenance program and is in compliance with the requirements of this directive paragraph (f)(1). This Agency does not take responsibility for the revision status of any Airplane Maintenance Manual held by the Airplane

Operator or other Agency with requirement of this directive paragraph (f)(2).

PILATUS

SB 21-009 Ref 213 dated 23 Nov 2007 concerning cabin pressure switches modification is accomplished at this time.

SB 25-033 Ref 208 dated 13 Jul 2007 concerning flight compartment instrument covers is accomplished by incorporation of kit # 500,50.12.309.

SB 30-010 Ref207 dated 24 May 2007 concerning modification of the OAT system is accomplished at this time by incorporation of kit P/N 500.50.12.310. SB 55-002R1 Ref 205 dated 30 May 2008 concerning incorporation of an improved vertical stabilizer shear plate and is corrected to be in compliance with the Bulletin Rev 1.

PIL 2008-02 'Inspection of carbon graphite panels' is accomplished per Pilatus Notice #17, Corrective Action paragraph.

Pilatus SL 123 dated 8 Sep 2008, concerning Starter Generator improvements is accomplished at this time. Installed an exchange starter generator P/N 23085-029 S/N P1015 and removed P/N 23085-025 S/N Y10145 per AMM 12-A-24-30-01-00A-920A-A. This unit is overhauled by FAA CRS G84R924X on W/O R21342 dated 28 Aug 2008. Other maintenance items addressed at this time

1) Installed new manufacture pilot and copilot seat back cushion supports P/N's 959.30.01.216.

2) Installed a new manufacture propeller low pitch warning switch P/N 973.10.11.101 per AMM 12-A-61-20-02-00A.

3) Performed Adjustment/Test of WOW proximity switches. Adjusted the R/H switch IAW AMM 12-A-32-10-00-00A.

4) Resealed the NLG shock strut per AMM 12-A-32-10-00-00A-904A-A.

DATE: 29 Dec 2008

5) Installed a new manufacture L/H MLG tire P/N 021-350-0 S/N 7362W00176 and removed S/N 2164W00093 per AMM 12-A-32-40-01-00A-920A-A.

6) Inspected inside of passenger door for water marks and corrosion per customer request. No defects noted per AMM 12-A-52-10-01-00A-920A-A.

7) Removed surface corrosion on body and end of left and right MLG actuator. Applied corrosion preventative Dinitrol AV 30 per AMM 12-A-20-40-00-00A-904A-A.

8) Resealed the L/H MLG shock absorber strut per AMM 12-A-32-10-00-00A-904A-A.

9) Serviced the hydraulic accumulator N2 per AMM 12-A-29-10-02-00A-901A-A.

10) Installed a new manufacture propeller beta ring follower carbon block per AMM 12-A-61-10-01-00A-920A-A.

11) Installed an exchange L/H outboard flap actuator P/N 978.73.20.309 S/N 1044and removed P/N 978.73.20.307 S/N 0647per AMM 12-A-27-50-03-00A-920A-A steps 2.2.2.3.3.4 & 3.5. This unit is overhauled by CH.145.0228 on W/O R22282 dated 22 Oct 2008, having 2023.5 total hours and 1825 total cycles in service.

(continued on the next page)

DATE: 29 Dec 2008	AIRCRAFT L	OGBOOK INSERT			Pg 3 of 3	Registration	on No		1000
	PILATUS	MODEL		C-12/45	_				C-1
N NUMBER N466SA 12) Installed an exchange L/H inboard flap act 22,2,3,3,4 & 3,5. This unit is overhauled by Cl	£145.0228 on W/O R22286 d	397and removed P/N 97 sted 22 Oct 2008, having	8.73.20.307 S/ 5058.8 total b 8.73.20.307 S/	N 0641per AMM 12-A	-27-50-03-00A-920A-A steps	and Remarks	policed 1	Signature	Certura
(3) Installed an exchange Nrth intons visp-wi- 22_3_3_4 & 3.5. This unit is overhanised by Cl 44) Installed an exchange RH outboard flap a 22_3_3_4 & 3.5. This unit is overhanised by Cl 45) Installed an exchange cabin differential pre required and returned to service by FAA CRS R 16) Installed an exchange cabin altitude pressure by FAA CRS RTIR464K on W/O W/O W/O W/O 17] Installed an exchange cabin altitude pressure by FAA CRS RTIR464K on W/O W/O 18] Repaired floor boards as per SRM 12-A-51- 19] Corrected the cause of incorrect displays on firm Symbol Generator were capped and stowe COS330 and resourced the SG input. Corrected 00A-90,1A-A. This unit is repaired by FAA CRS check flight no discrepancies noted. 20) Installed a new manufacture No. 4 passenge 21) Installed a new manufacture VII-1 and RVII MI. 23) Repaired scratches to the cockpit L/I side w 24) Installed new manufacture oxygen gauge c 25) Repaired as an amall puthole in the L/II wing in [Lut # 10-14647) per manufacture oxygen gauge c 25] Repaired as new manufacture oxygen gauge c 26] Installed a new manufacture oxygen gauge c 27] Corrected leaks at the polos a dilmenter. Tene 28] Corrected leaks at the polos a dilmenter. Tene 28] Corrected leaks at the polos a dilmenter. Tene 28] Corrected leaks at the polos a dilmenter. Tene 28] Corrected leaks at the polos a dilmenter. Tene 28] Corrected leaks at the polos a dilmenter. Tene 28] Corrected leaks at the polos a dilmenter. Tene 28] Corrected leaks at the polos a dilmenter of the 28] Corrected leaks at the polos the ine swivel per 28] Repaired the L/I MLIG Drisk line swivel per 28] Corrected leaks at the polos the ine swivel per 28] Corrected leaks at the polos the ine swivel per 29] Repaired the L/I MLIG Drisk line swivel per 29] Repaired the L/I MLIG Drisk line swivel per 29] Repaired the L/I MLIG Drisk line swivel per 29] Repaired the L/I MLIG Drisk line swivel per 29] Repaired the L/I MLIG Drisk line swivel per 29] Repaired the L/I MLIG Drisk line swivel per 29] Repaired the L/I MLIG Drisk line swi	1.145,0228 on W/O R22283 de trustator PN 978-73-20.309 SN 1.145,0228 on W/O R22280 de surre gauge PN 965.23.21.207 TIR 464K on W/O W/O 81560 e switch PN 965.23.21.533 S/ ted 16 May 3008. stare switch PN 965.23.21.533 S/ ted 16 May 3008. stare switch PN 965.23.21.533 S/ ted 16 May 3008. stare switch PN 965.23.21.534 S/ ted 16 May 3008. stare switch PN 965.21.21.534 1324 dated 15 May 2008. start placed for GNS530 and the 429.4 to B and B to A. Ins PRZR/93L on W/O AF5142 c seat back hydrolock PN HLI to window inner pane. PN 556.12 G actuator to wing attachment indow inner pane. ppillary tube PN 957.12.22.607 card decret boot, two feet from "switch PIN 582-10.A2BOC1, virting schematics." appearance on the EFIS displays. Mobel from pin 1 to pun 31, and satisfactory results. Primm PN 955.91.823 S/N 1 AMM 12-A-32-40-66-00A-90 or service/discrepancy items rath has been imported in accordingly per current Regulations of the 5521. Anworthinens Directions of the primary and proceed in according the current Regulations of the 5521. Anworthinens Directions and proceed in according the current Regulations of the 5521. Anworthinens Directions and the service of the proceed in according the current Regulations of the 5521. Anworthinens Direction and proceed in according the current Regulations of the 5521.	sted 22 Oct 2008, having the 1409 and removed PN 1409 and removed PN 1409 and removed PN 1500 and 1600	7,004.6 total in 7,704.6 total in 7,905.8 total herd S/N 10132. S/N 1013648 j ved S/N 10142 el 12BZ. of HDG and c ran to and para to and to an and an antipara to an and an antipara to an and an antipara to an and an anal an an anal an	sours and 829 10612 to \$28 N 6649per AMM 12-A-21- per AMM 12-A-21- per AMM 12-A-21- per AMM 12-A-21- per AMM 12-A-21- on the fuel page, the gre- lifel with the KMD850 \$SN 5828 and removed the GNS530 switch/ann \$44. A. A. g RAPCO Bootsaver D. g jumpers from pin3 to ults. 2 marker inputs on P80. in 29. Function tested in 12-A-34-25-01-00A-92 mined to be in airworthy.	A 27-50-03-00A-920A-A steps cs in service. 30-06-00A-920A-A. This unit is 30-06-00A-920A-A. This unit is 30-06-00A-920A-A. This unit is 30-08-00A-920A-A. This unit is 30-08-00A-A. Service Boot Repair kit, P/N 09-41705 pin2, pin 6 to pin 9, and pin11 to pin 30 of the KN40. Moved Inner marker arker receiver and displays per 30A-A. Scondition. The Aircraft and/or return to service. Pertinent details	Gold Medal Paint Co. FTG-Watking Co. Changed Pilatus PC-12 N4665A TO N55ZTC Per US.DOT ASIGMENT OF PEGISTRATION NUMBERS DATED QUEZENO.	N# 552TC Regregioman Chapage to KIDG TRAN 900 5/N 30 Prefire OF This aircraft was repaired Aviation Administration as INSPECTOR: June Front Ran	d and inspected ICW current reguli gd is approved for return to service of the service of the se	Date 1/17/6 N"Number Bosh Bosh arr (18 d

			Registration No	TEST TO A	
L&L Aviation LLC 11740 Airport Way 36F	Reg.# N552TC Model PC 12/45 S/N 443	Hour Meter Time: 1113.1 Aircraft Total Time: 1113.1	and Remarks aug 12, 2009	Signature	Certificate No.
per SB27-020 dated 01-30-09.	dated 03-30-09, Inspection of stick Bleed brake system and serviced w LN 90B GPS, Functional checks of Larry D. Qualls A&P 22288	systems were satisfactory.	AD 2009-14-13 Amdt 39-15963 Dathe requirement for insertion of Struck —Airworthiness Limitations, document dated 28 Jan 2009 into the Limitation maintenance program (e.g., maintenance per the directive paragraph (f) (currently in compliance with Structure Airworthiness Limitations, document dated 28 Jan 2009 with respect to the (f)(1)(i)(ii)&(2)(i).	octural and Component nent 12-B-04-00-00-00 ons section of the FAA ance manual) is accord 1) requirement. This aral and Components I at 12-B-04-00-00-00A	s Limitations 0A-000A-A, approved applished at this s aircraft is Limitations—

THE REAL PROPERTY.

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100	WESTERN AIRCRAFT

4300 5 KENNEDY ST., BOISE, IDAHO 83705-8132, U.S.A.

PHONE 208-338-1800

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AIRCRAFT LOGBOOK INSERT

AIRCRAFT MAKE PILATUS MODEL.

PC-12/45

N NUMBER N552TC HOBBS: 1175,3 LDGS: 828 .

Completed Annual Airframe and Engine inspections using Western Aircraft checklist Form MX-007 Rev 12 as per Pilatus PC-12/45/47 AMM Rev 19, dated 1 Mar 2009, section 12-A-05-20-20-00A-281A-A, and Pratt and Whitney Canada Model PT6A-67B Maintenance Manual, part number 3038336, Rev. 35, dated 17 Jul 2009, section 72-00-00, table 601 (minor, routine, and 100 hour inspection items). Accomplished the requirements of AMM 12-A-04-00-00-00A-000A-A. Oil sample shipped for SOAP analysis.

Chapter 4 & 5 Items addressed at this time

12 month flap actuator backlash inspections are accomplished per AMM12-A-05-10-20-00A-28 [A-A requirement. The results are:

EOB .	P/N 978.73.20.309	S/N 1409	Backlash .006	TTSN 1175.3	TCSN 828
RIB	978.73.20.309	0131	.007	1175.3	828
LOB	978 73 20 309	1044	.007	1175.3	828
LIB	978.73.20.309	1397	.006	1175.3	828

6 month airfull deicer surface treatment is accomplished by application of Age Master No. 1 per AMM12-A-05-10-20-00A-281A-A requirement.

- 12 month fire extinguisher inspection and contents check is accomplished per AMM12-A-05-10-10-00A-281A-A requirement,
- 12 month ELT inspection (S/N J092-0048) is accomplished per FAR 91-207 (d) (1) (2) (3) & (4). No discrepancy noted. Battery change is due 11-2013.
- 12 month emergency power supply capacity check PS855B S/N 3131 is accomplished per CMM 24-30-08 & L-3 SL 170. Capacity is greater than 100%. Reinstalled and tested per AMM 24-52-00.
- 12 month landing gear wheel bearing lubrication is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-12-20-06-00A-902A-A.
- 12 month WAI recommended landing gear shock strut hydraulic and nitrogen service check is accomplished per AMM 12-A-32-10-00-00A-902A-A and AMM 12-A-32-20-00-00A-902A-A
- 12 month/150 Hour flap screw actuator lubrication is accomplished per AMM12-A-05-10-20-00A-281A-A requirement & AMM 12-A-27-50-00-00A-902A-A.
- 12 month/ 200 Hour spark igniter inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement, AMM 12-A-74-20-00-00A-920A-A and EMM 74-20-00
- 12 month/ 400 Hour FWD battery (P/N 976.31/302) service is performed per AMM12-A-05-10-20-00A-281A-A requirement and Marathon CMM BA 24-34-00 Rev 3. The battery S/N 10103557 is deep cycled at this time. Installed one new cell P/N 0114-01C in the No. 4 position. Capacity check is 85%.
- 12 month/ 400 Hour AFT battery (P/N 976.31.302) service is performed per AMM12-A-05-10-20-00A-281A-A requirement and Marathon CMM BA 24-34-00 Rev 3 The battery S/N 10103554 is deep cycled at this time. Capacity check is 85%.
- 12 month/400 Hour propeller lubrication is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-12-20-05-00A-902A-A.
- 12 month: 500 Hour torque limiter check is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM12-A-71-00-00-00A-903A-A, Table 501)
- 12 month/ 500 Hour torque pressure and stick pusher torque transducer electrical connector cleaning is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-71-00-03-00A-250A-A.

(continued on the next page)

DATE: 18 Jan 2010 AIRCRAFT MAKE		AFT LOGBO	OK INSERT	MODEL		-12/45	Pg 2 0j 3
		CAL	443	HOBBS:	1175.3	_ LDGS:_	
NUMBER		complished per Al	MM12-A-05-10-2	0-00A-281A-A req	pairement and E	MNI /2-00-00	2. A. 22. 10.00-00A-313A-
2 month/ 600 Hour chip dete 2 month /1000 Hour autopile	at serve clutch check	are accomplishe	d per AMM12-A	05-10-20-00A-281	A-A requiremen	281A-A room	rement and AMM12-A-32
2 month /1000 Hour autopile 2 month /2000 Hour main la	nding gear axles and	bushings inspec	tion is accomplisi	hed per AMM 12-A	-03-10-20-007	-20114-14 respe	Action and the contract of
0-00-00A-313A-A	NEW			alished our AMMI	2. A.DE-00-00-0	A-A000A-A	
0-00-00A-313A-A. 2 month/3000 Hour horizon	tal stabilizer trim au	ral warning func	tion test is accom	burnen ber vereier			
requirement and AMM 12-A-2	7-40-01-00A-920A-/		20034-0000Rev	C Para 5.3.1.			
year RVSM conformity insi year RVSM conformity insi	pection is accomplish	ed per CMM 006	20034-0000Rev	Para 5.3.2			2000000
year RVSM conformity insp year oxygen bottle DOT by	pection is accomplish	polished per AMI	d 12-A-04-00-00	00A-000A-A and I	DOT-E-8162-18	150 requiremen	nt. Ref details below.
year oxygen bottle DOT hy	drostatic test is accor	application per AN	IM12-A-05-10-20	-00A-281A-A requ	airement and AN	dM12-A-35-2	0-00-00A-313A-A.
year passenger exygen mas	k inspections are acc	perational test is	accomplished pe	r AMM12-A-05-10	-20-00A-281A	A requiremen	t and AMM 12-A-28-40-0
year/ 1000 Hour fuel low ic	Act wat ming al second	Free Street Control of					

15 year exygen bottle DOT retirement is accomplished per AMM 12-A-04-00-00-00A-000A-A and DOT-E-8162-1850 requirement. Reference details below. 150 Hour elevator control cable tension check is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-27-50-00-00A-902A-A.

200 Hour starter generator brush wear check is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-24-30-01-00A-313A-A. 400 Hour generator No.2 Mod C brush inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-24-30-02-00A-313A-A. 600 Hour secondary heater thermal protection switches test is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-40-09-00A-903A-A

Airworthiness Directives and Service Bulletins addressed at this time

AD 2001-25-10 Amdt 39-12563 dated 25 Jan 2002 concerning flap system repetitive actuators backlash inspection is accomplished per the directive paragraph (d) (5) IAW AMM 27-50-03 as allowed by FAA AMOC dated 21 Apr 2003. The requirements of paragraph (d)(1)(2)(3) & (4) have been previously accomplished. The

next required inspection is due at 1775.3 ACTT. AD2009-08-11 Amdt 39-15884 dated 26 May 2009, concerning venting of the deice pressure regulator is determined to be N/A to MSN 443 per the directive paragraph (c).

Other maintenance items addressed at this time

- 1) Installed an exchange new manufacture standby altimeter P/N 5934PAD-3 A260 S/N 468777 and removed S/N 36887. Reference accompanying log entry.
- 2) Removed and reinstalled the KMC321 P/N 065-0086-17 S/N 2079 Installed a new backlight lamp per CMM P/N 006-05659-0003 Rev. 3 Dated Oct., 1993
- 3) Replaced No.1, 3, and aft LH masks and installed new fittings P/N's 8570-00.
- 4) Bled the hydraulic brake system per AMM 12-A-32-40-00-00A-902A-A.
- 5) Downloaded the engine EIS card and sent to Trend Group.
- Installed an exchange new manufacture R/H wing strobe lamp P/N 03-2201-0002 S/N 40350 and removed S/N 28673.
- 7) Removed and reinstalled the elevator per AMM 12-A-55-20-01-00A-920A-A. Repaired a minor dent on the L/H per Elevator Repair Procedure RM08221-Trailing Edge. Reapplied coatings and performed a static balance procedure per SRM 12-A-51-60-02-00A-276A-A.
- 8) Installed a new Diode TB 115-07 P/N 971.42.31.880.
- 9) Installed new engine oil cooler door actuator rod P/N 571.10.12.353 and 571.10.12.354 using IPC 12-A-79-20-00-02A-941A-A as reference.

(continued on the next page)

DATE : 18 Jan 2010 AIRCRAFT MAKE	AIRCRAFT LOGBOOK			Pg 3 of 3			OFFI
N NUMBER 10) Repaired a pinhole in L/H wine		MODEL 443 HOBBS: er repair kit instructions.		828	and Remai	MAKE: Pilatus MODEL: PC12/45 S/N: 443 N NUMBER: N552TC I certify that the altimeter, automatic pressure altitude reporting and static system tests have been performed per CFR 14 part 43 Appendix E (b)(1)(2) to see the static system tests have	
11) Ins aircraft is altered by install Kelner Pilatus Center, Inc., 2001 K. Center, Inc., 2012 K. Center, Inc., Linc., 2012 K. Center, Inc., Linc., 2013 K. Center, Inc., Liptup Land List have been revised. Concerning this alteration has been 12: Installed PM Research Protectic 13: Installed PM Research Protectic 13: Installed a new manufacture ox manufacture ox manufacture ox MLG door seals 15: Installed new L/H and R/H ailer 17: Removed and reinstalled the Flidated 4 Jan 2010. 18: Installed a new manufacture cor 19: Installed a new manufacture car 20: Fabricated one passenger seat ba (a) (1) (ii) by Flame-Out Inc. DER 12: Installed a loaner cockpit overtherpaired by FAA CRS HSRR811E or 23: Installed an exchange stick push repaired by FAA CRS ZP8R849Y or 23: Installed an exchange stick push repaired by FAA CRS ZP8R849Y or 23: Installed and cleaned three pneum A-30-10-00-00A-903A-A. 24: Adjusted the roll null on the A/I Corrected miscellaneous minor inter MAINTENANCE RELEASE: Leruity if MAINTENANCE RELEASE: Leruity if a concerning the content of the corrected miscellaneous minor inter MAINTENANCE RELEASE: Leruity if	llation of a Propeller Feathering Inhibit Sy, Celner Place, Thunder Bay, On P78 679, ctions for Continued Airworthiness docum Il Supplement VKPC-FIS-001 dated 10 O This installation results in on change 10 th submitted to the FAA Registration Brancion Kit PM199. This bottle is also hydrostatic tested PN's 557.24.12.215, 557.24.12.216 and 57 imit placard P/N 511.23.12.028. ron and rudder L/H and R/H warning plac ight System Display CPU, P/N FD200CP ekpit instrument glare shield P/N 531.12.1 go door maximum weight placard P/N 51 ase shroud using Kydex LLC #52000 P-1. Herb Reed II test ID 11779. Painted to mad control panel P/N 972.81.32.102 S/N on W/O 12288 dated 8 Sep 2009. her computer P/N 975.44.23.104 S/N 1201 on W/O B9107 dated 29 Sep 2009.	stem in accordance with STC SAO Canada. Instructions for continue tent number VKPC-PC12-F104. I tt 2006 has been incorporated inte e aircraft's weight and balance doe at this time. The next hydrostatic S7.24.12.212. ards P/N 511.21.12.038 per AMM U-7 S/N 7007112. This unit is rep 2.035. 1.34-12.147 per AMM 12-A-11-00 060", Kydex 100. This covering h tch existing interior color. JWPIL147 and removed S/N 336 159 and removed S/N 111495 per 2-A-30-10-02-00A-901A-A. Satin till - END with an Annual Inspection and is detere	animothiness are conti- Dated 16 Nov 2006. The Date	ained in V. Kelner Pilatus required V. Kelner Pilatus ating Handbook. The aircraft 37 dated 5 Jan 2010 -920A-A. This unit has a 5. 1A-A. R276B on W/O JM1410 25.853 (a) Appendix F Part I I-00A-920A-A. This unit is DA-920A-A. This unit is saccomplished per AMM 12-dition. The Aircraft and/or rum to service. Pertinent details		been performed per CFR 14 part 43 Appendix E (b) (1)(2) to comply with CFR 14 part 91.411(a)(1) requirements. Inspected the aircraft per RVSM 12 month and 24 month requirements in accordance with CMM 006-20034-0000Rev F Para 5.3.1 and Para 5.3.2, with the exception of Para5.3.2(c). #1 Altimeter P/N: 70256N02D02 S/N: 07090849D Feet: 35000 #2. Altimeter P/N: 70256N02D02 S/N: 07090850D Feet: 35000 Standby Altimeter P/N: 5934PAD-3 A260 S/N: 468777 Feet: 35000 Details of this inspection are on file at this repair station under W.O. # 6798. #2 Western Aircraft 4300 S. Kennedy St. Boise, ID 83705 CRS FE6R532N Signature Date: 18 January 2010 MAKE: Pilatus MODEL: PC12/45 S/N: 443 N NUMBER: N552TC The ATC Transponder tests and inspections required by FAR 91.413 were performed this date and found to comply with Part 43, Appendix F. #1 – Model: KT70 P/N 06601144-0201 S/N: 30360 P/N: 30360 P/N: 4300 S. Kennedy St. Boise, ID 83705 CRS FE6R532N Details of this inspection are on file at this repair station under W.O. # 6798. #1 – Model: KT70 P/N 06601144-0201 S/N: 30360 P/N: 30572 Details of this inspection are on file at this repair station under W.O. # 6798. **Western Aircraft** 4300 S. Kennedy St. Boise, ID 83705 CRS FE6R532N Signature Date: 18 January 2010 P/N 06601144-0201 S/N: 30572 Details of this inspection are on file at this repair station under W.O. # 6798. **Western Aircraft** 4300 S. Kennedy St. Boise, ID 83705 CRS FE6R532N Signature Date: 18 January 2010 P/N 06601144-0201 S/N: 30572 Details of this inspection are on file at this repair station under W.O. # 6798.	Certificate No



WWW A Crosmotor Agreement	4300 S KENNEDY ST., BOISE, IDAHO 83705-8132, U.S.A. PHONE 208-338-1800
	AIRCRAFT LOGBOOK INSERT
Date: 19 April, 2010	
AIRCRAFT MAKE: PILA	ATUS MODEL: PC-12/45 N NUMBER: N552TC S/N: 443 HOBBS: 1184.1
Performed the following mai (i) Reinstall the repaired original theck good IAW 12-A-31-10-	internance. ginal overhead panel, P/N 972.81.32.102, S/N JWPIL336 per Pilatus IAW 12-A-31-10-01-00A-920A-A. Ops -00-00A-903A-A Ops checked good. Loaner unit removed same P/N, S/N JWPIL147.
	- END - Aircraft and/or component identified above was repaired and/or inspected, per current Regulations of the Federal Aviation Administration, and (ound to be
	inworthy with respect to the work performed, subsequently approved for return to service this date: 19 April, 2010. Pertinent details are on file at this agency under Shop Work Order No. 6905.
	/ , , , , ,
Signati	For FAA CLETTIFIED REPAIR STATION NO FEORSON Michael L Flamm
Total to	
Carry Forward	

Date: April 5, 2010

AIRCRAFT LOGBOOK INSERT

AIRCRAFT MAKE PILATUS

MODEL PC-12/45

TOTAL TIME 1181.4

N NUMBER 552TC

S/N 443

CYCLES 832

Airworthiness Directives and Service Bulletins addressed at this time:

AD 2010-1-03 Amdt 39-16159 dated Jan. 20, 2010, concerning replacement of defective Fire Fighting Enterprises Ltd. fire extinguishers, is complied with by replacing the fire extinguisher per the directive paragraph (g)(1). Installed an Amerex Model C352TS 2 ½ Lb Halon 1211 S/N X-83564 in the same location as the original and removed FFE S/N 038836. The weight change is negligible. The bottle manufacture date is 2009.

MAINTENANCE RELEASE: I certify that this Aircraft has been inspected in accordance with a 100 Hour Inspection and is determined to be in airworthy condition. The Aircraft and/or component identified above was inspected and/or repaired, per current Regulations of the Federal Aviation Administration, and is subsequently approved for return to service.

Signature of authorized Person Tauf 7.7 Juny TA 1508577 4/5/10

L&L Aviation LLC 11740 Airport Way 36F Broomfield, Co. 80021 Reg. # N552TC

Model PC 12/45 S/N 443

Hour Meter Time: 1191.2

Aircraft Total Time: 1191.2

April 23, 2010

Removed right main tire assembly removed tire, clean and visual inspected wheel. Installed new Michelin tire S/Nor batch number 9155W00122 P/N 021-350-0. Serviced with nitrogen, and reinstalled assembly back onto right main gear.

Larry D. Qualls A&P 2228865



4300 S KENNEDY ST_BOISE, IDAHO 83705-8132, U.S.A.

PHONE 208-338-1800

DATE: 11 Feb 2011 AIRCRAFT MAKE

AIRCRAFT LOGBOOK INSERT

PILATUS MODEL PC-12/45

N NUMBER N552TC HOBBS: 1250.9 LDGS: 859est Completed Annual Airframe and Engine inspections using Western Aircraft checklist Form MX-007 Rev 14 as per Pilatus PC-12/45/47 AMM 02049 Rev 22, dated

15 Apr 2010, section 12-A-05-20-20-00A-281A-A, and Pratt and Whitney Canada Model PT6A-67B Maintenance Manual, part number 3038336, Rev. 36, dated 30 Apr 2010, section 72-00-00, table 601 (minor, routine, and 100 hour inspection items). Accomplished the requirements of AMM 12-A-04-00-00-00A-000A-A. Oil sample shipped for SOAP analysis

Chapter 4 & 5 Items addressed at this time

12 month flap actuator backlash inspections are accomplished per AMM12-A-05-10-20-00A-281A-A requirement. The results are:

Location ROB	P/N 978.73.20.309	S/N 1409	Backlash 006	TTSN 1250.9	TCSN 859	
RIB	978.73.20.309	0131	.006	1250.9	859	
LOB	978.73.20.309	1044	.007	1250.9	859	
LIB	978.73.20.309	1397	007	1250.9	859	-

6 month airfuil deicer surface treatment is accomplished by application of Age Master No. 1 per AMM12-A-05-10-20-00A-281A-A requirement

- 12 month fire extinguisher inspection and contents check is accomplished per AMM12-A-05-10-10-00A-281A-A requirement
- 12 month ELT inspection (S/N J092-0048) is accomplished per FAR 91-207 (d) (1) (2) (3) & (4). No discrepancy noted. Battery change is due 11/2013,
- 12 month emergency power supply capacity check PS855B S/N 3131 is accomplished per L-3 CMM 24-30-08 and SL-170F. Capacity is greater than 100%. Reinstalled and tested per AMM 24-52-00
- 12 month landing gear wheel bearing lubrication is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-12-20-06-00A-902A-A.
- 12 month WAI recommended landing gear shock strut hydraulic and nitrogen service check is accomplished per AMM 12-A-32-10-00-00A-902A-A and AMM 12-A-32-20-00-00A-902A-A
- 12 month/150 Hour flap screw actuator lubrication is accomplished per AMM12-A-05-10-20-00A-281A-A requirement & AMM 12-A-27-50-00-00A-902A-A.
- 12 month/ 200 Hour spark igniter inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement, AMM 12-A-74-20-00-00A-920A-A and EMM 74-20-00
- 12 month/ 400 Hour Fwd battery (P/N 976.31.302) service is performed per AMM12-A-05-10-20-00A-281A-A requirement and Marathon CMM BA 24-34-00 Rev
- 3. The battery S/N 10103557 is deep cycled / serviced at this time. Capacity check is greater than 100%
- 12 month/400 Hour Aft battery (P/N 976.31,302) service is performed per AMM12-A-05-10-20-00A-281A-A requirement and Marathon CMM BA 24-34-00 Rev 3. The battery S/N 10103554 is deep cycled / serviced at this time. Capacity check is greater than 100%.

(continued on the next page)

Registration No.

AIRCRAFT MAKE	PII	LATUS		MODEL	PC	-12/45		
DATE : 11 Feb 2011 AIRCRAFT MAKE N NUMBER	N552TC miter check is accomp	S/N	443				: 859est	

- requirement and AMM 12-A-71-00-03-00A-250A-A.
- 12 month/ 600 Hour chip detector inspection is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and EMM 72-00-00
- 12 month /1000 Hour autopilot servo clutch checks are accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM12-A-22-10-00-00A-313A-A.
- 12 month /2000 Hour main landing gear axles and bushings inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM12-A-32-10-00-00A-313A-A
- 12 month/3000 Hour propeller feathering switches tests are accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-76-10-02-00A-903A-A
- 12 month/3000 Hour horizontal stabilizer trim aural warning function test is accomplished per AMM12-A-04-00-00-00A-000A-A
- requirement and AMM 12-A-27-40-01-00A-920A-A.
- I year RVSM conformity inspection is accomplished per CMM 006-20034-0000Rev C Para 5.3.1.
- 2 year Generator #1 and #2 grounding point inspections are accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-24-30-00-00A-901A-A
- 2 year CAWS CACU clock battery replacement is accomplished per 12-A-05-10-10-00A-281A-A requirement & AMM 12-A-31-50-03-00A-920A-A.
- 2 year AHRS system calibration & stand-by compass checks are accomplished per AMM 12-A-05-10-20-00A-281A-A requirement, AMM 12-A-34-25-00-00A-903A-A and AMM 12-A-34-21-00-00A-903A-A
- 2 year/1500 Hour pressurization outflow valve and safety valve cleaning is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-30-01-00A-250A-A
- 2 year/1500 Hour pressurization system adjustment and test is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-30-00-00A-903A-A
- 200 Hour starter generator brush wear check is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-24-30-01-00A-313A-A.
- 200 Hour P3 air filter cleaning is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and EMM 73-10-07.
- Airworthiness Directives and Service Bulletins addressed at this time
- AD 2001-25-10 Amdt 39-12563 dated 25 Jan 2002 concerning flap system repetitive actuators backlash inspection is accomplished per the directive paragraph (d) (5) IAW AMM 27-50-03 as allowed by FAA AMOC dated 21 Apr 2003. The requirements of paragraph (d)(1)(2)(3) & (4) have been previously accomplished. The next required inspection is due at 1850.9 ACTT.
- SB 31-008 Ref 237 dated 22 Sep 2009 concerning modification to the systems which are controlled by the overhead panel is accomplished at this time.
- SB 32-023 Ref 254 dated 15 Dec 2009 concerning NLG steering rod end retention improvement is accomplished at this time.

(continued on the next page)

DATE : 11 Feb 2011		AIRCRA	FT LOGBO	OK INSERT(co.	ntinued)		D 1 44			
AIRCRAFT MAKE		PILATUS M				-12/45	Pg 3 of 3			
N NUMBER	N552TC	S/N		HOBBS:						
Other maintenance items addr	essed at this time		110	nobbs:_	1250.9	LDGS:	859est .			
Cleuned Airframe fuel filter p	er AMM 12-A-28-20-	01-00A-920A-A.								
Performed functional check o Performed post maintenance	f the ailerons per AM?	M 12-A-27-10-00-	00A-903A sectio	n 3.2.1-3.3.5 No de	feets noted					
Performed post maintenance This aircraft is absend at the	pitot static leak checks	per AMM 12-A-3	4-11-10-00A-90	3A-A	neets noted.					
					and installation	of souls				
A01376CH dated 10 Jan 2001 fannifin main landing gear whe	which is held by Park	er Hannifin Corpor	ration, 1160 Cen	ter Road, Avon. OH	44011 This a	lteration come	nt wheels and brakes per ST			
lannifin main landing gear whe 41 Rev. H dated 25 Aug. 2008	els and brakes P/N's 4	10-424 & 30-244 (c	conversion kit No	. 199-241) in accord	lance with Park	meration cons	ists of installing Parker			
41 Rev. H dated 25 Aug. 2008 is required FAA Approved Air	See FAA Form 337 c	dated 30 Oct 2009	for the details of	this conversion. The	aircraft Pilots	Operation Us	istaliation manual No IM19-			
ne required FAA Approved Air pdated to reflect the changes in	plane Flight Manual S	upplement Docum	ent No. 199-241	AFMS dated 22 June	2009 with the	operation ria	ndbook is revised to include			
pdated to reflect the changes nated 10 January 2001. Installed	sade. Instructions for o	continued airworth	iness are contain	ed in the FAA Appro	ved Aimlane	Maintenance A	farance and equipment list			
ated 10 January 2001. Installed 244S00428. The brake system	a new manufacture L	/H MLG tire P/N (021-350-0 S/N 0:	244S00477 Installed	a new manufa	eture PALM	G time Day 621			
244S00428. The brake system If wheel assembly removed-P	is serviced as per AM	M 32-40-00 with t	he ops and leak	hecks ok. Part and S	erial #'s are a	s follows:	Ci tire P/N 021-350-0 S/N			
H wheel assembly removed- P	7N 3-1543 & SN 0774	LH wheel assemb	ly installed- PN	40-424 & SN 2164/2	187	a rollows.				
					2186.					
 This aircraft is altered at this SA02742NY dated 19 Nov 200 	time by the removal o	f the factory-origin	al Hartzell Prope	fler, and the installat	ion of an MT-	Propeller S.ble	de seco II esse			
\$A02742NY dated 19 Nov 200 natural composite propeller IAV	, neid by MT-Propell	er Entwicklung Gr	nbh, Flugplatzstr	asse 1, 94348 Artig.	Germany The	s alteration of	ide propener per STC			
satural composite propeller IAV additional details. The aircraft	MI-Propeller Install	lation Instruction I	Occ. No. E-1824,	Rev. 2 dated 10 Nor	v 2010 Refer	ence EAA For	misists of installing a 5-blade			
dditional details. The aircraft Rev. No. 1 dated 25 Feb 2010.	Pilot's Operating Hans	fbook is revised to	include the requ	ired FAA Approved	Airplane Fligh	Manual Sun	m 337 dated 10 Feb 2010 for			
Rev. No. 1 dated 25 Feb 2010. AT-Propeller Doc. No. E-1825	Weight and balance r	ecords have been t	apdated to reflect	the changes made	Instructions for	r continued air	nement Doc. No. E-1823,			
AT-Propeller Doc. No. E-1825 (J1660, Propeller installed- M	, original Issue dated i	2 Nov 2009. Part	and Serial #'s ar	e as follows: Propel	ler removed-	Hartrell DALL	Worthiness are contained in			
o 1909, Propener installed- M.	I-Propelley P/N MTV.	37.1.N C E D (D)	CHIPSON AND AND AND			tentacil P/P4 I	1C-E4A-3D/E10477K S/N			
A Installed a way was for	hydraulic low pressure	21-1-14-C-1-14(1-)	CFR260-65a, S/	N 100793			The state of the s			

DA

7) Serviced the oxygen level per the AMM 12-A-12-10-05-00A-902A-A. Performed a leak test of the oxygen system with all passenger and crew masks disconnected 8) Installed a new manufacture NLG wheel P/N 3-1501 S/N 1259 and removed S/N 0374 per CMM 32-47-72 page block 701 and AMM 12-A-32-40-02-00A-920A-A. Corrected miscellaneous minor interior and exterior service/discrepancy items. - - END- -MAINTENANCE RELEASE: I certify that this Aircraft has been inspected in accordance with an Annual Inspection and is determined to be in airworthy condition. The Aircraft and/or component identified above was inspected and/or repaired, per current Regulations of the Federal Aviation Administration, and is subsequently approved for return to service. Pertinent details are on file at this agency under Shop Work Order No. 2742. Airworthiness Directives have been checked through BW 2011-01 Date. 11 Feb 2011. Signature of authorized Person Ronnie D. Bortz

For FAA CERTIFIED REPAIR STATION NO. FE6R512N

978.73.20.309 Part Nr. 1409 Serial Nr.

Description Actuator, Flap, Saba

The repair/overhaul record for the part, product or appliance identified has been reviewed. Based on an audit of EASA Form 1 tracking number 146149 dated 10/22/2008 and supporting documentation, this article is approved for return to service as of date

Documentation on file at Pilatus Business Aircraft, Ltd., 11755 Airport Way, Broomfield, CO 80021, PH (303) 465-9190. Print Name Jerry L. Frank

Date 2 April 2011 A&P Certificate 523276450

978.73.20.309 Part Nr. Serial Nr. 1397

of release referenced on EASA Form 1.

Actuator, Flap, Saba Description

The repair/overhaul record for the part, product or appliance identified has been reviewed. Based on an audit of EASA Form 1 tracking number 146148 dated 10/22/2008 and supporting documentation, this article is approved for return to service as of date of release referenced on EASA Form 1. Documentation on file at Pilatus Business Aircraft, Ltd., 11755 Airport Way, Broomfield, CO 80021, PH (303) 465-9190.

Print Name Jerry L. Frank

A&P Certificate 523276450

Date 2 April 2011

978.73.20.309 Part Nr.

Actuator, Flap, Saba Description

0131

Serial Nr. The repair/overhaul record for the part, product or appliance identified has been reviewed. Based on an audit of EASA Form 1

tracking number 146150 dated 10/22/2008 and supporting documentation, this article is approved for return to service as of date of release referenced on EASA Form 1. Documentation on file at Pilatus Business Aircraft, Ltd., 11755 Airport Way, Broomfield, CO 80021, PH (303) 465-9190.

Print Name Jerry L. Frank

Date 2 April 2011

A&P Certificate 523276450

	AIRCRAFT LOG	MAKE: Pilatus MODEL: PC12/45	
Part Nr. 978.73.20.309 Serial Nr. 1044	Description Actuator, Flap, Saba	S/N: 443 N NUMBER: N552TC	Certificate N
The repair/overhaul record for the part, produc	t or appliance identified has been reviewed. Based on an audit of EASA Form 1	and been performed per CFR 14 part 43 Appendix E (b)(1)(2) to comply with CFR 14 part	Commedie
tracking number 146145 dated 10/22/2008 a	and supporting documentation, this article is approved for return to service as of date	01.411(a)(1) requirements. Inspected the aircraft per RVSM 12 month and 24 month	
of release referenced on FASA Form 1	atticle is approved for return to service as of data	requirements in accordance with CMM 006-20034-0000Rev F Para 5.3.1 and Para 5.3.2,	
Documentation on file at Pilatus Business Airc	raft, Ltd., 11755 Airport Way, Broomfield, CO 80021, PH (303) 465-9190.	with the exception of Para5.3.2(c).	
Signature		#1 Altimeter P/N: 70256N02D02 S/N: 07090849D Feet: 35000 .	
	Print Name Jerry L. Frank	#2 Altimeter P/N: 70256N02D02 S/N: 07090850D Feet: 35000 .	
A&P Certificate 523276450	Date 2 April 2011	Standby Altimeter P/N: <u>5035P-P</u> S/N: <u>46887</u> Feet: <u>35000</u>	
Part Nr. 972.81.32.102	Description OVERHEAD PANEL	Details of this inspection are on file at this repair station under W.O. # 8787.	
Serial Nr. JWPIL336	OVERHEAD PANEL	Western Aircraft	
The repair/overhaul record for the part, produc	t or appliance identified has been reviewed. Based on an audit of EASA Form 1	4300 S. Kennedy St. Boise, ID 83705 CRS FE6R532N -	
tracking number UWI-AAT-RA 100305 01 date	d 3/5/2010 and supporting documentation, this article is approved for return to	4500 S. Reiniedy St. 4500 S.	
service as of date of release referenced on EA	SA Form 1.	Signature Steven Jacquell Date: May 8, 2012.	
Socumentation on the at Phatus Business Airci	raft, Ltd., 11755 Airport Way, Broomfield, CO 80021, PH (303) 465-9190.	For FAA CERTIFIED RAPAIR STATION NO. FE6RS32N	
Signature Musica Starting		V	
	Print Name Dennis Phillips		
A&P Certificate 2118896	Date 6 April 2011	MAKE BU A MODEL POINTS	
Part Nr. 978.73.20.016		MAKE: Pilatus MODEL: PC12/45	
Serial Nr. 108923	Description FCWU	S/N: 443 N NUMBER: N552TC	
The repair/overhaulance		The ATC Transponder tests and inspections required by FAR 91.413 were	
tracking number EMC-AAT-00 57397 details	ct or appliance identified has been reviewed. Based on an audit of EASA Form 1	performed this date and found to comply with Part 43, Appendix F.	
do of date of release referenced on FACA F	and the second s	#1 - Model: <u>KT70</u> P/N <u>06601144-0201</u> S/N: <u>30360</u>	
Documentation on the at Pilatus Business Airc	n 1. craft, Ltd., 11755 Airport Way, Broomfield, CO 80021, PH (303) 465-9190.	#1 – Model: <u>KT70</u> P/N <u>06601144-0201</u> S/N: <u>30572</u> .	
Signature 1/1 1/1.		Details of this inspection are on file at this repair station under W.O. # 8787.	
Signature Chennes Mullips	Print Name Dennis Phillips	Western Aircraft	,
A&P Certificate 2118895	Date 8 April 2011	4300 S. Kennedy St. Boise, ID 83705 CRS FE6R532N	
Carry Forward	- April EVII	- Disc, in division of the factorial	
		Signature Steven J. Cowell Date: May 8, 2012.	
		For FAA CERTIFIED REPAIR STATION NO. FE6R532N	



4300 S KENNEDY ST., BOISE, IDAHO 83705-8132, U.S.A.

PHONE 208-338-1800

DATE: May 8, 2012	AI	RCRAFT	LOGB	OOK INS	10-12/43	
AIRCRAFT MAKE	PI	LATUS		MODEL	PC	-12/45
N NUMBER	N552TC	S/N	443	HOBBS:		LDGS: 876

Completed Annual Airframe and Engine inspections using Western Aircraft checklist Form MX-007 Rev 15 as per Pilatus PC-12/45/47 AMM 02049 Rev 24, dated Dec 2011, section 12-A-05-20-20-00A-281A-A, and Pratt and Whitney Canada Model PT6A-67B Maintenance Manual, part number 3038336, Rev. 37, dated 8 Oct 2010. section 72-00-00, table 601 (minor, routine, and 100 hour inspection items). Accomplished the requirements of AMM 12-A-04-00-00-00A-000A-A. Oil sample

Chapter 4 & 5 Items addressed at this time

12 month flap actuator backlash inspections are accomplished per AMM12-A-05-10-20-00A-281A-A requirement. The results are:

ROB RIB LOB LIB	978.73.20.309	5/N 1409	Backlash .007	TTSN 1284.3	TCSN
RIB	978.73.20.309	0131	.008	1284.3	938
LOB	978.73.20.309	1044	.007	1284.3	938
LIB	978.73.20.309	1397	.009	1284.3	938

- 6 month airfoil deicer surface treatment is accomplished by application of Age Master No. 1 per AMM12-A-05-10-20-00A-281A-A requirement
- 12 month fire extinguisher inspection and contents check is accomplished per AMM12-A-05-10-10-00A-281A-A requirement.
- 12 month ELT inspection (S/N J092-0048) is accomplished per FAR 91-207 (d) (1) (2) (3) & (4). No discrepancy noted. Battery change is due 11/2013.
- 12 month emergency power supply capacity check is accomplished per L3 SL-170 Rev F, CMM 24-30-08. Capacity is satisfactory. Reinstalled and tested per
- 12 month landing gear wheel bearing lubrication is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-12-20-06-00A-902A-A 12 month WAI recommended landing gear shock strut hydraulic and nitrogen service check is accomplished per AMM 12-A-32-10-00-00A-902A-A and AMM 12-
- 12 month/150 Hour flap screw actuator Jubrication is accomplished per AMM12-A-05-10-20-00A-281A-A requirement & AMM 12-A-27-50-00-00A-902A-A. 12 month/ 200 Hour spark igniter inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement, AMM 12-A-74-20-00-00A-920A-A and EMM
- 12 month/ 500 Hour torque limiter check is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM12-A-71-00-00-00A-903A-A, Table 501)
- 12 month/500 Hour torque pressure and stick pusher torque transducer electrical connector cleaning is accomplished per AMM 12-A-05-10-20-00A-281A-A
- 12 month/ 600 Hour chip detector inspection is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and EMM 72-00-00.
- 12 month /1000 Hour autopilot servo clutch checks are accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM12-A-22-10-00-00A-313A-A. (continued on page 2)

AD 2001-25-10 Amdt 39-12563 dated 25 Jan 2002 concerning flap system IAW AMM 27-50-03 as allowed by FAA AMOC dated 15 Jul 2011. The required inspection is due at 1884.3 ACTT.	m repetitive actuators backlash inspection is accomplished per the directive paragraph (d) (5) a requirements of paragraph (d)(1)(2)(3) & (4) have been previously accomplished. The next
SB 34-029 Ref 270 dated 30 Mar 2011 concerning cockpit display wirin, SB 23-008 Ref 278 dated 28 Sept 2011, concerning correction of oxyger when compared to the proper in the SB. Theocephia	mask microphone socket wiring is not applicable due to wiring configuration being different

AIRCRAFT LOGBOOK INSERT(continued)

12 month/3000 Hour horizontal stabilizer trim aural warning function test is accomplished per AMM12-A-04-00-00-00A-000A-A

12 month /2000 Hour main landing gear axles and bushings inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM12-A-32-10-

12 month/3000 Hour propeller feathering switches tests are accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-76-10-02-00A-

4 year/ 1000 Hour fuel low level warning system operational test is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM 12-A-28-40-00-

5 year/ 2000 Hour flight control cable inspections are accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-27-00-00-00A-313A-A.

5 year duct over temperature switch inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-60-05-00A-920A-A. 10 year/ 15000 flight/ 11000 hour NLG torque tube retirement is accomplished per 12-A-04-00-00-00A-A requirement. Installed P/N 532.50.12.064 tube. 10 year/ 11000 Hour wing to fuselage attachments inspection is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-57-00-00-00A-10 year/ 11000 Hour cockpit side windows inspection/check is accomplished per AMM 12-A-05-10-20-00A-281A-A and AMM 12-A-56-11-02-00-A-313A-A. 200 Hour starter generator brush wear check is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and AMM 12-A-24-30-01-00A-313A-A.

3 year passenger oxygen mask inspections are accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM12-A-35-20-00-00A-313A-A

5 year over temperature switch inspection is accomplished per AMM12-A-05-10-20-00A-281A-A requirement and AMM 12-A-21-40-06-00A-903A-A.

MODEL

PILATUS

N552TC

- prints in the SB. The problematic symptoms described in the SB are not present in this aircraft
- SB 24-017R2 Ref 143 dated 20 Sept 2001 concerning removal of the original dual Ni-Cad storage batteries and installation of a dual lead acid batteries is accomplished at this time. Installed new FWD Lead acid battery P/N RG-380E/44, S/N 40490121 Installed new AFT Lead acid battery P/N RG-380E/44, S/N 40490161.

Pg 2 of 3

PC-12/45

HOBBS: 1284.3 LDGS: 876

Other maintenance items addressed at this time

requirement and AMM 12-A-27-40-01-00A-920A-A.

DATE: May 08, 2012

AIRCRAFT MAKE

N NUMBER

00-00A-313A-A

00A-903A-X

- 1) Re-lamped the tail nav light PN 972.87.87.184 per AMM 12-A-33-43-01-00A-920B-A.
- 2) Installed three new chafe strips P/N's 557.24.12.215 and 557.24.12.216 on the L/H MLG door.
- 3) Installed a new L/H MLG tire P/N 021-350-0 S/N 1266S00166 per CMM 40-424.

Airworthiness Directives and Service Bulletins addressed at this time

4) Lowered the cockpit overhead liner and re-secured the pilot and co-pilot air gasper ducting. 5) Serviced the breathing oxygen supply level per the AMM 12-A-12-10-05-00A-902A-A.

(continued on page 3)

200 Hour P3 air filter cleaning is accomplished per AMM 12-A-05-10-20-00A-281A-A requirement and EMM 73-10-07.

DATE: May 8, 2012		AFT LOGBO		CASE CONTROL C			Pg 3 of 3	
AIRCRAFT MAKE		LATUS		MODEL	PC-	12/45		
N NUMBER	N552TC	S/N	443	HOBBS:		LDGS:	876 .	
us unit is repaired by UK. 14: Installed a new L/H engine I 11A-A. Re-lamped the AHRS 2 & V. 10) Verified seats #3 and #4 au orrected miscellaneous mino IANTENANCE RELEASE: 1 omponent identified above was in file at this agency under Shop W.	temperature control val 5 00870 on W/O 3698; neated inlet lip sealing: WX 2 indicators P/N's re in the correct "modif interior and exterior se- certify that this Aircraft has spected and/or repaired, po-	lve P/N 6031C000- 233-001000 31054 flange and new seal 6839. fied position As Peservice/discrepancy as been inspected in acc er current Regulations Airworthiness Directive	901 S/N 1310 an 9864 dated 10 M ling cord and new er Report No. LE items END ordance with an An of the Federal Aviat es have been checke	d removed P/N 5825 ar 2012. rivets per AMM 12 -1627 and installed p mual Inspection and is di- tion Administration, and	2-A-78-10-01-00 clacards on the fetermined to be in subsequently a Date_May 8, 201	0A-920A-A ar loor panels to airworthy condi	identify their locations, tion. The Aircraft and/or arn to service. Pertinent details are	
SO)		TOLFANCE	SKITITED KE(AII	STATION NO. FEOR	532N			
WESTERN A								
WESTERN A A Graverwich Auto	Group! Company	4300 S KENNEDY	Y ST., BOISE, IDA	.HO 83705-8132, U.S	A	PHONE 2	208-338-1800	
A Greenwich Aerol	Group! Company	4300 S KENNEDY	Y ST., BOISE, IDA		A	PHONE 2	208-338-1800	
DATE : 15 Feb 2011 AIRCRAFT MAKE	Group' Company A I PI	4300 S KENNEDY IRCRAF	Y ST., BOISE, IDA		A ERT PC	-12/45		
	Group Company Ali PI N552TC	4300 S KENNEDY IRCRAF	Y ST., BOISE, IDA	HO 83705-8132, U.S OOK INS	A ERT PC	-12/45	208-338-1800 	

	Registration N	0	10			
and Remarks		and west	Signature	Certificate No.		
			-			
	Boots West	Solution				
	Patch-less De-ic					
Model Serial	PC12/45 443 N 552TC	Work Order Hour Meter	8/10/12			
Registration	N 552TC	Date	8/10/12			
filled pin ho functional / Specification Details on fi	eumatic surface de-icer les, repaired splits and o leak test, and applied w n A-001, FAA Approve ile under work order ide Approved rep FAA Certificated Repai	cuts, resealed edge ax in accordance v d on Form 8110-3 entified above.	s, performed with Process dated 5/26/96.			
	Thunderbolt and Lig	htfoot Marketing l	ne.			

	Registra	ition N	lo					_			_					
nd Remarks											Sig	gna	ture		Ce	rtificate No.
2	-02 s/n 386. See FAA pair Services. Trim system	÷.	spunod	inches aft of di	inch-pounds	spunod	spunod		spunod	inches aft of di	inch-pounds	spunod	spunod	d in roved for		
MARK T I O N 65260 - (480) 948-240 ottember 28, 201 1/45 S/N 443 3 LNDS 969	LOG ENTRY tuator p/n 129-1-110 9-1-1100-02 s/n 825, 831126 by Derco Re the horizontal stab. Performed autopilot	3.18" aft of divider wa inguration	6562.90	235.91	1548252.74	3,358.10	9,921.00	nfiguration	6627,90	236.59	1568093.19	3,293.10	9,921.00	nance was performed this aircraft is app	1111	
LANDMARK A A A D N A A A A A A A A A	Removed the horizontal stabilizer actuator pin 129-1-110-02 s/n 386. Installed overhaulida stabilizer actuator pin 129-1-110-02 s/n 386. from 8130-3 dated 9/18/2012, WO# R31126 by Derco Repair Services TTSN 2020. GSN 1939. Performed the horizontal stab. Trim addictment test, no problems noted. Performed autopilot system adjustment, no problems noted.	Installed #7 position jump seat at 160.18" aft of divider wall. Six seat configuration	New Aircraft Empty Weight:	New Aircraft Empty Weight CG:	New Aircraft Moment:	New Aircraft Useful Load:	Maximum TO Weight:	Eight seat configuration	New Aircraft Empty Weight:	New Aircraft Empty Weight CG:	New Aircraft Moment:	New Aircraft Useful Load:	Maximum TO Weight:	certify that the above stated maintenance was performed in accordance with the current FAR's and this aircraft is approved for eturn to service. W.O. P29-12-00004	ANDMARK AVIATION	

			AFT LOG Recorder	
Date	Aircraf This Date	Total	Reading	Maintenance
NDMA A T A S280 (48)	N552TC Date: October 19, 2012 Model # PC12/45 S/N 443 ACTT: 1348.3 LNDS 969 AIRFRAME LOG ENTRY CW Task 21-40/55, 500 hour Cold air unit re-oil.	I certify that the above stated maintenance may be accordance with the current FAR's and this aircraft is approved for return to service. W.O. P29-12-00011 FAA CRS# LN3R663L FAA CRS# LN3R663L		



AIRFRAME PC-12/45

12/14/2012 S/N: 443

N552TC

HOBBS: 1430.0

C/W 100 Hour Inspection IAW 12-A-05-20-10-00A-281A-A Checklist.

C/W 150 Hour Flap Actuator Lube.

C/W 500 Hour Water Separator Sock Cleaning. Replaced sock with new P/N 959.90.20.143.

Designation No.

C/W 500 Hour Cabin Heat Exchanger Matrix Inspection/Check with no discrepancies noted.

Removed hydraulic pressure switch P/N 973.81.14.304, S/N 2099 and installed new P/N 973.81.14.307, S/N 0353. Ops and leak

checks good and CAWS light is extinguished. Replaced L/H Main Wheel valve core with new, Serviced to 60 PSI and leak checked with no defects noted.

Replaced tail nav bulbs with qty (2) new P/N 972.87.87.184.

Replaced beacon lamp with new P/N AHQH28V150WL20C (972.87.87.333).

Replaced beta brush block with new P/N A3074.

Performed functional checks as required with no defects noted.

Airworthiness Directive compliance checked through Revision 2012-24.

Work performed IAW Pilatus PC12 Document No. 02049 Rev. 25 and Pratt & Whitney MM 303836 Rev 38.

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH A 100 HOUR INSPECTION SELECTED BY THE OWNER/OPERATOR UNDER PROVISIONS OF FAR 91.409(f)(3) AND WAS DETERMINED TO BE IN AN AIRWORTHY CONDITION AND IS APPROVED FOR RETURN TO SERVICE. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 4023611.

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260 480-991-0900

Perorder



AIRFRAME PC-12/45

2/23/2013 S/N: 443

N552TC

HOBBS: 1485.0

C/W Starter/Generator Brush Inspection per item # 24-30-01-00A-513A-A. Brush wear is 65% remaining.

C/W #1 & #2 Generator ground point inspection per item # 24-30-00-00A-901A-A.

C/W Cabin Pressurization System Test per item # 21-30-00-00A-903A-A.

C/W Cabin Pressurization Controller Filter Cleaning per item # 21-30-01-00A-250A-A.

C/W AAHARS Calibration on #1 & #2 AHRU per item #12-A-34-25-00-00A-040A-A.

C/W Standby Compass Swing per item # 12-A-34-21-00-00A-903A-A. Updated compass card.

Replaced CAWS Battery with new P/N 972.81.52.050 per item # 31-50-03-00A-920A-A. Battery manufacture date is 12/11 with a

replacement date of 2/21/2015.

Performed functional checks as required with no defects noted.

Work performed IAW Pilatus PC12 Document No. 02049 Rev. 26 and Pratt & Whitney MM 303836 Rev 38.

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED/REPAIRED IN ACCORDANCE WITH CURRENT FARS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 4024095.

480-991-0900

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730

7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260

Carry Forward

Registration No.



Page 1 of 2 N552TC AIRFRAME PC-12/45

5/9/2013 S/N: 443

HOBBS: 1517.6 AFTT: 1517.6 LANDINGS: 1046

C/W Annual Inspection IAW 12-A-05-20-20-00A-281A-A Checklist.

C/W 12 month/ 400 hour Capacity Check of Main Batteries. Removed P/N RG380E/44, S/N 40490121 & S/N 40530967 and reinstalled after cap check. No defects noted.

C/W 12 month Capacity Check of Emergency power Supply Battery. Removed P/N 501-1712-02, S/N 3131 and re-installed after cap check by Duncan Avionics under WO# 275NA dated 5/3/2013.

C/W 12 month Main & Nose Wheel Bearings Lubrication IAW 12-A-12-20-06-00A-902A-A.

C/W 12 month / 200 hour Main Landing Gear & Bushes Inspection IAW 12-A-32-10-00-00A-313A-A with all bushes found to be in limits.

C/W 12 month / 500 hour Torque Limiter Test IAW 12-A-71-00-00-00A-903H-A.

C/W 12 month / 500 hour Oil Pressure Transducer, Engine Torque and Stick Pusher Torque Transducer Electrical Connector Cleaning IAW 12-A-71-00-03-00A-250A-A.

C/W 72 month/ 6,000 hour Mild Corrosive Environment Inspection IAW 12-A-05-10-20-00A-281A-A with no corrosion noted.

C/W 24 month/ 2,000 hour Compressor Recovery Performance Wash IAW 12-a-71-00-00A-250A-A.

C/W 150 Hour Flap Actuator Lube IAW 12-A-27-50-00-00A-902A-A.

C/W 12 month Portable Fire Extinguisher Weight Check.

C/W 12 month/ 600 hour Chip Detector Continuity Inspection.

C/W 600 hour Fuel Pump Inlet Screen Cleaning & Fuel Pump Outlet Filter Replacement. Replaced filter assembly with new P/N AN6235-3A.

C/W 12 month ELT Inspection IAW FAR91.207 (d). battery Due Date is: November 2013.

C/W 12 month/ 1,000 hour Servo Mount – pitch, roll, yaw, Clutch Inspections IAW 12-A-22-10-00A-313A-A.

C/W 6 month Airfoil De-Icers Top Coat by application of Age master No. 1.

C/W A.D. 2012-26-16 "Chp. 4 Limitations" by verifying Chp. 4 airworthiness limitations 12-A-04-00-00-00A-000A-A is part of the current aircraft maintenance manual. This is terminating action.

Replaced R/H MLG axle lock bolt with new P/N AN3C21 and new nut P/N MS17825-3.

Removed over voltage protection device for external power receptacle P/N 988.21.16.101, NSN and installed new P/N 988.21.16.104. S/N 130458.

Replaced placards on emergency exit door with new P/N 511.21.12.042 & P/N 511.21.12.040.

Replaced chafe tape on L/H engine cowling support bulkhead with new P/N 021200-24343. Replaced R/H MLG up bumper with new P/N 957.24.12.075. Registration No.



Page 2 of 2 N552TC

PC-12/45

5/9/2013 S/N: 443

Re-secured cargo door handle trim using new velcro strips.
Replaced pitch trim limits placard with new P/N 511.24.12.048.
Performed functional checks as required with no defects noted.
Airworthiness Directive compliance checked through Revision 2013-09.

Work performed IAW Pilatus PC12 Document No. 02049 Rev. 26 and Pratt & Whitney MM 303836 Rev 38.

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH AN ANNUAL INSPECTION SELECTED BY THE OWNER/OPERATOR UNDER PROVISIONS OF FAR 91.409(f)(3) AND WAS DETERMINED TO BE IN AN AIRWORTHY CONDITION AND IS APPROVED FOR RETURN TO SERVICE. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 4024799.

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE , AZ 85260 480-991-0900

	47.2-3		
RIVE ATTEM		4	



AIRFRAME PC-12/45

8/5/2013 S/N: 443

N552TC

HOBBS: 1644.2 AFTT: 1644.2 LANDINGS: 1143

C/W 100 Hour Inspection IAW 12-A-05-20-10-00A-281A-A Checklist.

C/W 200 Hour Cap. Check of #1 & #2 Lead Acid Batteries. Removed P/N RG-380E/44, S/N 40530967 and S/N 40490121 & re-

installed after can check. C/W 200 Hour Starter/Generator Brushes Inspection. Noted approx. 70% brush life remaining.

C/W 150 Hour Flap Actuator Lube IAW 27-50-00-00A-902A-A.

C/W 12 month/ 3,000 Hour Horizontal Stabilizer Trim Check IAW 12-A-27-40-00-00A-903A-A.

Removed R/H main wheel assembly. Disassembled, cleaned, inspected with no defects noted. Removed tire P/N 40-424, S/N Unknown and re-assembled using new tire P/N 40-424, S/N 2025S00294 Re-packed wheel bearings and re-installed on aircraft.

Serviced tire to 60psi. Work performed IAW Cleveland AWBCMM0001-9 CMM and 12-A-32-40-01-00A-920A-A.

Performed flap adjustment/test procedure IAW 12-A-27-50-00-00A-903A-A.

Performed functional checks as required with no defects noted.

Airworthiness Directive compliance checked through Revision 2013-15.

Work performed IAW Pilatus PC12 Document No. 02049 Rev. 26 and Pratt & Whitney MM 303836 Rev 38.

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH A 100 HOUR INSPECTION SELECTED BY THE OWNER/OPERATOR UNDER PROVISIONS OF FAR 91.409(f)(3) AND WAS DETERMINED TO BE IN AN AIRWORTHY CONDITION AND IS APPROVED FOR RETURN TO SERVICE. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 4025634.

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260 480-991-0900

AIRFRAME Pilatus PC12/45 9/13/2013 S/N: 443

N552TC

N552TC

HOBBS: 1673.0

Removed #1 AHRS P/N 124210-2022, S/N Unknown and installed serviceable P/N 124210-2022, S/N 1083. Performed functional checks as required with no defects noted. Work performed IAW Pilatus MM 02049 Rev 026.

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED/REPAIRED IN ACCORDANCE WITH CURRENT FARS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 94025872.

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 480-991-0900 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260

AIRFRAME PC-12/45

11/26/2013 S/N: 443

HOBBS: 1735.6 AFTT: 1735.6

Serviced nitrogen on hydraulic accumulator to 53 PSI. Performed functional checks as required with no defects noted. Work performed IAW PC12 Document No. 02049 Rev. 26.

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED/REPAIRED IN ACCORDANCE WITH CURRENT FARS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 4026559..

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260 480-991-0900

	DESCRIPTION AND A PAGE NO			Registration No	
N552TC	AIRFRAME PC-12/45	12/13/2013 S/N: 443	-1-	CHICATIN ARCHITAGA	
C/W 200 Hour/ 3 mor 40490121 & re-installec C/W 60 month Cabin C/W 60 month Cabin C/W 60 month Cabin C/W 144 month Crew P/N 1-10-870301, qty WO#40011 dated 12/1 C/W 6 month airfoil d Removed L/H main w Unknown and re-asser bearings and re-installe 40-01-00A-920A-A. Applied edge seal on R Replaced R/H nav lens Replaced R/H lip heat Replaced L/H lip heat Replaced L/H lip heat Replaced upper tail nav Performed functional c Airworthiness Directiv Work performed IAW I CERTIFY THAT T INSPECTION SELI AND WAS DETERM TO SERVICE. PERT WORK ORDER 4026 IN EXECUTIVE AIRCI	thion IAW 12-A-05-20-10-00A-281A-A Checklist. Inth Cap. Check of #1 & #2 Lead Acid Batteries. Removed P/N RG-2 d after cap check. Altitude Switch Test (Installed) IAW 12-A-21-30-07-00A-903A and fo Differential Pressure Switch Test (Installed) IAW 12-A-21-30-08-00A- and Passenger Seat belt Overhaul IAW 12-A-05-10-10-00A-281A-A re (1) P/N 1-10-875301 and qty (6) P/N 5-02-024723MBLRevA by C&M 12/2013. le-icers top coat. Reference West Coast Wash Station log entry for deta theel assembly. Disassembled, cleaned, inspected with no defects noted inbled using new tire P/N 301-125-005, S/N M12298505 and O-Ring F d on aircraft. Serviced tire to 60psi. Work performed IAW Cleveland A A/H I.B. Horizontal stabilizer de-ice boot. Is with new P/N 972-87-87.187. Inlet tube with new P/N 530-24-12-124. Is almp with new P/N 972-87-87.184. Ichecks as required with no defects noted. It compliance checked through Revision 2013-24. Pilatus PC12 Document No. 02049 Rev. 26 and Pratt & Whitney MM ITHIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE ECCTED BY THE OWNER/OPERATOR UNDER PROVISION MINED TO BE IN AN AIRWORTHY CONDITION AND IS A TINENT DETAILS ARE ON FILE UNDER THIS FAA APPRO	und to be within limits. 903A-A and found to be within limits. placement schedule. Installed qty (1) I Marine Aviation Services under ils. Removed tire P/N 021-350-0, S/N //N MS28775-268. Re-packed wheel WBCMM0001-9 CMM and 12-A-32- 303836 Rev 39. E WITH A 100 HOUR IS OF FAR 91.409(p(3) PPROVED FOR RETURN DVED REPAIR STATION,	- assembly with new P/N 968.29.13. noted. Work performed IAW Pilan - I CERTIFY THAT THIS AIRC FARS AND IS APPROVED FO - ARE ON FILE UNDER THIS: INSPECTOR	RAFT HAS BEEN INSPECTED/REPAIRED IN A R RETURN TO SERVICE FOR THE WORK PERFAA APPROVED REPAIR STATION, WORK ORD	CCORDANCE WITH CURRENT FORMED. PERTINENT DETAILS ER 4027011.

ESTOT THE ASSESSMENT AND ASSESSMENT

AIRFRAME PC-12/45

5/13/2014 S/N: 443

HOBBS: 1870.9 AFTT: 1870.9 LANDINGS: 1328

C/W Annual Inspection IAW 12-A-05-20-20-00A-281A-A Checklist.

C/W 12 month Corrosion Protection Inspection.

Page 1 of 2

N552TC

C/W 600 hour Secondary Heater Thermal Switches Inspection.

C/W 500 hour Air Conditioning Cold Air Unit lubrication.

C/W 12 month/ 1,000 hour Servo Mount - pitch, roll, yaw, Clutch Inspections IAW 12-A-22-10-00A-313A-A.

C/W 12 month/ 600 hour Capacity Check of Main Batteries. Removed P/N RG380E/44, S/N 40490121 & S/N 40530967 and reinstalled after cap check. No defects noted.

C/W 400 hour Brush Check on #2 Generator, Approx, 80% life remaining on brushes.

C/W 12 month Capacity Check of Emergency power Supply Battery. Removed P/N 501-1712-02, S/N 3131 and re-installed after cap check by Duncan Avionics under WO# 2YY3A dated 5/5/2014.

C/W 12 month/ 150 hour Flap Actuator Lube IAW 12-A-27-50-00-00A-902A-A.

C/W 12 month/ 1,000 hour Flap Actuator backlash Check.

C/W 12 month Main & Nose Wheel Bearings Lubrication IAW 12-A-12-20-06-00A-902A-A.

C/W 12 month / 200 hour Main Landing Gear & Bushes Inspection IAW 12-A-32-10-00-00A-313A-A with all bushes found to be in limits.

C/W 12 month / 500 hour Torque Limiter Test IAW 12-A-71-00-00-00A-903H-A.

C/W 12 month / 500 hour Oil Pressure Transducer, Engine Torque and Stick Pusher Torque Transducer Electrical Connector Clearing IAW 12-A-71-00-03-00A-250A-A.

C/W 12 month/ 3,000 hour Horizontal Stabilizer Trim Runaway Aural Warning Check.

C/W 12 month Portable Fire Extinguisher Weight Check.

C/W 1,000 hour Starter/Generator Overhaul. Removed P/N 23085-029, S/N P1015 and re-installed after overhaul by Southwind Aviation Supply under WO#25960 dated 5/8/2014.

C/W 500 hour Heat Exchanger Matrix Inspection.

C/W 500 hour Water Separator Condenser Cleaning.

C/W 500 hour Vapor Cycle Compressor Drive Belt & Compressor Motor Check/ Inspection. Serviced with 2.0 lbs R-134a freon per 12-A-21-50-00-00A-903B-A. Current Hobbs reads 453.9.

C/W AD 2001-25-10 "Flap System Failure" recurring 600 hour inspection IAW 12-a-27-50-03-00A-313B-A with no defects noted. Next Due in 600 hours at 2470.9 AFTT.

C/W 12 month ELT Inspection IAW FAR91.207 (d). Battery Due Date: 11/2020.

C/W 24 month altimeter and transponder checks as required by CFR 91.411 and 91.413 and was found to comply with CFR Part 43 Appendixes E and F. The altimeter(s) were tested to 30.000 feet.



Page 2 of 2 N552TC

5/13/2014 S/N: 443

Removed co-pilot O2 mask P/N 9NC-10-06-121, S/N 95587 & sent for repair. O2 mask to be re-installed at a later date. Co-Pilot seat is not to be occupied during flights at cabin pressure altitudes above 15,000 feet per FAR 91.211 (a)(3) until mask is re-installed.

Replaced R/H lower nav light lamp with new P/N 972.87.87.184.

Replaced chafe tape on L/H & R/H flaps with new.

Removed L/H & R/H landing lights P/N GE4596 and installed Whelen direct replacement LED lamps P/N 11-08292.

PC-12/45

Removed taxi light P/N GE4626 and installed Whelen direct replacement LED lamp P/N 11-08293.

Bucked qty (2) rivets under pilot side passenger seat rail IAW Pilatus SRM Chp. 51-40-01.

Performed functional checks as required with no defects noted.

Airworthiness Directive compliance checked through Revision 2014-09.

Work performed IAW Pilatus PC12 Document No. 02049 Rev. 26 and Pratt & Whitney MM 303836 Rev 38.

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED IN ACCORDANCE WITH AN ANNUAL INSPECTION SELECTED BY THE OWNER/OPERATOR UNDER PROVISIONS OF FAR 91.409(f)(3) TO MAD WAS DETERMINED TO BE IN AN AIRWORTHY CONDITION AND IS APPROVED FOR RETURN SERVICE. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 4027399 & 94025872.

	INSPECTOR:	
CUTIVE	AIRCRAFT MAINTER	MANOR DALL

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260 480-991-0900

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AIRFRAME Pilatus PC12/45

7/1/2014

S/N: 443

704-4-1

N552TC

Installed a Dual Garmin G600 Primary Flight Display (PFD) and Multifunction Display (MFD) System IAW STC SA02153LA-D. Issued 7/25/2008 and Garmin installation manual 190-00601-06, Re. L, 10/2013. See revised equipment list for P/N's and S/N's

Installed a Garmin GTN 650 and a 750 Navigational system IAW STC SA02019SE-D, Issued 3/18/2011 and Garmin installation manual 190-01007-A3. Re. 7, 12/2013. See revised equipment list for P/N's and S/N's Removed Bendix/King TCAS 1 traffic system and installed rack and wiring provisions for a Garmin GTS800 traffic system

IAW Doc. # 190-00993-4 Rev. 5, 6/2012. See revised equipment list for P/N's and S/N's. Removed composite spinner and installed new aluminum unit P/N B523-P-1059-AL, NSN.

Removed standby attitude gyro and installed a new Mid-Continent SAM MD302 IAW installation manual 9017782. Rev. F. 3/2014. See revised equipment list for P/N's and S/N's.

Removed GMA 340 P/N 011-00401-10; S/N 96264505 and installed serviceable loaner P/N 011-00401-10; S/N 96265007, Ground Operational check satisfactory.

Installed new boom/oxygen switches P/N MTC-206N. Complied with 24 month altimeter and transponder checks as required by CFR 91.411 and 91.413 and was found to comply with CFR Part 43 Appendixes E and F. The altimeter(s) were tested to 35,000 feet.

Interior removal and reinstallation work accomplished IAW Doc. 02049 MM Rev. 27.

INSPECTOR:

N552TC

AC Hobbs: 1890.9

THIS AIRCRAFT WAS INSPECTED AND/OR REPAIRED IN ACCORDANCE WITH CURRENT FAA REGULATIONS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTURENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION WORK ORDER: 94027936& 94028002.

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 25260 480-991-0900

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Signature

Certificate No.

AIRFRAME Pilatus PC12/45 AC Hobbs: 1904.8

7/10/2014 S/N: 443

Installed a Garmin GDL 69 XM Satellite Receiver IAW Garmin Installation manual 190-00355-02, Rev. K, 4/2011 and Garmin STC SA01487SE-D: issued 12/20/2004

Interior removal and reinstallation work accomplished IAW Doc. 02049 MM Rev. 27.

THIS AIRCRAFT WAS INSPECTED AND/OR REPAIRED IN ACCORDANCE WITH CURRENT FAA REGULATIONS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION WORK ORDER: 94028002.

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260 480-991-0900

l A	Aircraft Time Recorder			Signal	Certificate No.
N552TC HOBBS: 1917.5	AIRFRAME Pilatus PC12/45	7/22/2014 S/N: 443	and Remarks	ERCUTTY ARCAN THUMBEANCE AIRFRAME	
190-01007-A3 Rev. 7. Replaced abrasion tape on nos to match. Tape to be applied a Replaced qty (3) bulbs on land 00-00A-040A-A with no defect Work performed LAW Pilatus I CERTIFY THAT THIS A FARS AND IS APPROVED ARE ON FILE UNDER TO INSPECT EXECUTIVE AIRCRAFT.	ding gear indicators with new P/N 6839. Performed operation cts noted. MM 02049 Rev 027. AIRCRAFT HAS BEEN INSPECTED/REPAIRED IN D FOR RETURN TO SERVICE FOR THE WORK PER THIS FAA APPROVED REPAIR STATION, WORK OR	form L/H IB leading edge and painted area al checks of landing gear IAW 12-A-32-60- ACCORDANCE WITH CURRENT RFORMED. PERTINENT DETAILS DER 94028132. N E27R1730	THIS AIRCRAFT WAS INSPECTED A APPROVED FOR RETURN TO SERV FAA APPROVED REPAIR STATION INSPECTOR: INSPECTOR:	Pilatus PC12/45 429 DFCS speed as required IAW Garmin IM 190-0060 ry IAW KFC 325 IM 006-00659-0012, Rev. 12, 8/2006. Doc. 02049 MM Rev. 27 and above references. AND/OR REPAIRED IN ACCORDANCE WITH CURREN	NT FAA REGULATIONS AND IS FAILS ARE ON FILE UNDER THIS
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Parts under W Work perform I CERTIFY FARS AND ARE ON FI	p heat timer P/N 4E316 7/O R14-01885 Dated 7- ned, IAW Pilatus PC-12/ THAT THIS AIRCRA IS APPROVED FOR F LE UNDER THIS FA INSPECTOR:	Pilatus 3-2 S/N U273. And 31-14. Ops check no /45 AMM 30-10-03-0 AFT HAS BEEN IN RETURN TO SERVA APPROVED RE IENANCE FAA AI	installed overhauled P/N 4E3 rmal 0A-920A-A. No discrepancies SPECTED/REPAIRED IN VICE FOR THE WORK PE PAIR STATION, WORK OF	S/N: 443 163-2 S/N U0236. By Ametek B&S Aircraft noted. I ACCORDANCE WITH CURRENT RFORMED. PERTINENT DETAILS RDER 4028412.	- C/W 100 Hour Inspection IAW 12 C/W 3 month/ 200 Hour Cap. Ch 40490121 & re-installed after cap c C/W 60 month O2 Bottle Hydrost inspection. Reference Cylinder Rec C/W 6 month Airfoil De-Icer top Replaced tail nav lights lamps with Performed functional checks and I Airworthiness Directive compliant Work performed IAW Pilatus PCI I CERTIFY THAT THIS AIR INSPECTION SELECTED B AND WAS DETERMINED T TO SERVICE. PERTINENT WORK ORDER 4028626. INSPECTO EXECUTIVE AIRCRAFT MM	2-A-05-20-10-00A-281A-A Checklist. eeck of #1 & #2 Lead Acid Batteries. Remo eheck. tatic Inspection. Removed P/N 1274469, S, qualification Report dated 9/30/2014 by Ai Coat. Boots were stripped, a coat of agema enew P/N 1982SP. leak checks as required with no defects note ee checked through Revision 2014-19. 12 Document No. 02049 Rev. 28 and Pratt. CRAFT HAS BEEN INSPECTED IN A THE OWNER/OPERATOR UNDE TO BE IN AN AIRWORTHY CONDIT DETAILS ARE ON FILE UNDER TH	/N ALT2803339 and re-installe dant Fire, DOT Registration #I ster was applied and waxed with ed. & Whitney MM 303836 Rev 41 ACCORDANCE WITH A 10 ER PROVISIONS OF FAR 91 ION AND IS APPROVED FILE FAA APPROVED REPA	d after hydrostatic 3080. Serviced Oxygen. n no defects noted. 0 HOUR .409(f)(3)
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AIRFRAME 552TC Pilatus PC12/45 S52TC OBBS: 2018.8 semoved ED402 EFIS Screen P/N 066-03125-2500, S/N 3388 and installed Tested P/N 066-03125-2500, S/N U3671 by Honeywell atternational under WO# AP3373 dated 9/26/2014. Performed operational checks IAW Bendix King IM 006-00698-0007 Rev. 9 with a defects noted. Work performed IAW Pilatus MM 02049 Rev 028 and above references. Vork performed IAW Pilatus MM 02049 Rev 028 and above references. CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED/REPAIRED IN ACCORDANCE WITH CURRENT ARS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 94028643. INSPECTOR: EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION, WORK ORDER 640-991-0900 EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730	Date	Aircraft Time	Recorder	Maintenance	and Remarks		Signature	Cermicale 140.
Total to	552TC OBBS: 2018.8 emoved ED462 EF International under W o defects noted. Vork performed IAV CERTIFY THAT FARS AND IS APF IRE ON FILE UN EXECUTIVE AIR	FIS Screen P/N 066-03125-2500, S/N 3388 and in WO# AP3373 dated 9/26/2014. Performed opera W Pilatus MM 02049 Rev 028 and above reference T THIS AIRCRAFT HAS BEEN INSPECTE PROVED FOR RETURN TO SERVICE FOINDER THIS FAA APPROVED REPAIR STAINSPECTOR:	enstalled Tested P/N 066-031: ational checks IAW Bendix K es. ED/REPAIRED IN ACCO R THE WORK PERFORN ATION, WORK ORDER 9 D REPAIR STATION E27	S/N: 443 25-2500, S/N U3671 by Honeywell Ging IM 006-00698-0007 Rev. 9 with RDANCE WITH CURRENT MED. PERTINENT DETAILS 14028643.	HOBBS: 2103.1 Downloaded and cleared EIS memo A. Note: Battery Expiration Date: 2/ Replaced R/H main tire with new P. Performed ops and leak checks as re Work performed IAW Pilatus MM 0 I CERTIFY THAT THIS AIRCE FARS AND IS APPROVED FOR ARE ON FILE UNDER THIS F INSPECTOR: EXECUTIVE AIRCRAFT MAIN	Pilatus PC12/45 ory card. Replaced EIS battery with new P/1/28/2023. /N 850T06-3, S/N 42692371 IAW Pilatus equired with no defects noted. 12049 Rev 029. RAFT HAS BEEN INSPECTED/REP/R RETURN TO SERVICE FOR THE VEAL APPROVED REPAIR STATION, VEA	12-A-32-40-01-00A-920A-A. MIRED IN ACCORDANCE W WORK PERFORMED. PERTI WORK ORDER 4029463. R STATION E27R1730	

Pilatus PC12/45

1/20/15 S/N: 443

AIRFRAME

N552TC HOBBS: 2104.3

Touched up paint on propeller leading edges to match.

Repaired broken wire to toilet in reference to Pilatus 12-A-25-21-00-00A-040A-A. Work performed IAW Pilatus MM 02049 Rev 029 and above references.

I CERTIFY THAT THIS AIRCRAFT HAS BEEN INSPECTED/REPAIRED IN ACCORDANCE WITH CURRENT FARS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION, WORK ORDER 4029463.

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 480-991-0900 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260

N 552TC Make: Pilatus Model: PC-12/45

Serial #: 443

Page 1 of 1

Airframe Log Entry

Date: 10/17/14 Hobbs: 2030.0

LANDMARK AVIATION

for CRS 9PER215C

5500 - 44th 5t Building 403 Grand Rapids, Ml. 49512

* Troubleshot landing gear system to bad relay K601 pn:974.09.26.211 sn:N633. Removed failed relay and replaced with new relay pn:974.09.26.211 sn:N980 and performed operational check of landing system with no defects noted.

I certify that the aircraft, aircraft engine, propeller, or appliance identified above was repaired and or inspected in accordance with current FAA regulations and was found airworthy with respect to the work performed and is approved for return to service.

Signed: Steven J. Norris

Date: 10/17/14

WO# \$66- 14-61028

Registration No._

LANDMARK 14700 North Airport Drive • Scottsdale, AZ 85260 • (480) 948-2400

> N552TC Date: October 30, 2014 Model # PC12/45 S/N 443 ACTT: 2051.8 LNDS 1456 **AIRFRAME LOG ENTRY**

C/W the following tasks:

12-A-32-20-06-00A-313A-A. 2000 Hr Ultrasonic Inspection of the Nose Landing Gear Drag Link .

See Canyon State Inspection Field Inspection Report dated 10-29-14 CSI JOB# AC14-1111. No Defects Noted.

I certify that the above stated maintenance was performed in accordance with the current FAR's and this aircraft is approved for return to service. W.O. P29-14-00275

FAA CRS# LN3R663L

Certificate No.

Signature



N552TC Date: February 18, 2015 1 of 2 Model # Pilatus PC-12 S/N 443 ACTT: 2115.1

PT6A-67B S/N: PCE-PR0051 ETT 2115.1 HC-E4A-3D S/N: HJ-923 PTT: 2115.1

AIRFRAME LOG ENTRY C/W manufacturers 100 hr checklist IAW PC-12 MM 5-20-00

y I OI WUIL

C/W the following inspections: 100 hr airframe lubrication 12m/150 hr Flap Drive arm in- situ Inspection/check. 200hr Starter/Generator Brush wear check 3/4 life remaining

IAW PC 12 AMM 24-30-00. 200Hour/3month Lead acid battery cap check 24m/2000hr Pressurization Outflow Valve and safety valve cleaning IAW PC-12 MM 21-30-00 1000hr Fuel cross vent &Outward check valve inspection IAW

PC12 MM 28-10-00. 48m Generator1 & 2 ground point inspection, no defects noted IAW MM 24-20-00 48m Caws battery replacement with new P/N 972.81.32.050

IAW PC 12 MM 31-50-03 48m Standby magnetic compass swing IAW PC12 MM 6m Age Master Application IAW MFR's instructions

Serviced oxygen IAW PC-12 MM 12-10-00 Replaced Beta block with new P/N A-3074 Ran engine, check good IAW PC 12 MM 76-10-00



N552TC Date: February 18, 2015

2 of 2

Removed and replaced lower deice lip with new P/N 530.24.12.090 IAW MM 78-10-00

Removed and Installed new Hydraulic Pressure Switch P/N

9738114307 Bled hydraulic system, cycled landing gear, Operation check good no leaks IAW PC 12 MM 29-10-03

I certify that this AIRCRAFT has been inspected in accordance with an 100 hour inspection and was found to be in an Airworthy Condition. W.O. P29-15-00310

LANDMARK AVIATION

FAA CRS# LN3R663L

		RCRAFT LOG			Registration No		
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NS52TC NG Hobbs: 2116.4		AIRFRAME Pilatus PC12/45			AIRFRAME Pilatus PC12/45		4/22/2015 S/N: 443
IAW Garmin 2. All work according to the same care that approved for reference approved relationship to the same care that approved relationship to the sam	opilot GDU 620 fan plate assen manual 190-00601-10, Rev. complished IAW Doc. 02049 M.S. INSPECTED AND/OR REFETURN TO SERVICE FOR TEPAIR STATION WORK ORD INSPECTOR: EXECUTIVE AIRCRAFT MAIN 7920 E. GREENWA	1, 2/2014. Operational check MM Rev. 29 and above reference PAIRED IN ACCORDANCE VIEW WORK PERFORMED. PEDIR: 94028402.	satisfactory. nces. WITH CURRENT FAA REGU RTINENT DETAILS ARE OF	2. All work accomplished IAW I THIS AIRCRAFT WAS INSPECTEL APPROVED FOR RETURN TO SER FAA INSPECTOR:	ystem software to version 7.00 IAW Carmin SSB 15200601-00, Rev. L, 3/2015 were supplied to operator. Doc. 02049 MM Rev. 29 and above references. D AND/OR REPAIRED IN ACCORDANCE WITH CURVICE FOR THE WORK PERFORMED. PERTINENT APPROVED REPAIR STATION WORK ORDER: 94	RRENT FAA REGULATION DETAILS ARE ON FILE 030196.	ONS AND IS
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		AIRC	RAFT LOG		Ser.				
Date	Aircraft This Date	Time	Recorder Reading	342			Registration No	(TENTAL CONT.	
	Brought Forward		Redding	Maintenance	_	DUNCAN	N552TC Tempus Aircraft Sales & Se	Pilatus PC-12/45 rvice S/N: 443	June 16, 2015
			EXECUTIVE AIRCRAFT MANAGEMENTS			Performed ELT Inspection	on on Kannad 406AF ELT (P/N	N: S1821502-02 S/N: J092-0048) IAV Operational check normal. No defects n	W Kannad Service Letter SI
N552TC AC Hobbs: 2	2128.6		AIRFRAME Pilatus PC12/45		3/30/2015 S/N: 43	Battery S1820506-01 Battery Expiration Date : 1 Beacon ID # ADCC018A5			
1. Re 2. All	paired copilot windshie work accomplished IA\	ld sealant as requ W Doc. 02049 MM	ired IAW PPG SIL-172121- I Rev. 29 and above referen	001, Rev. B, 1/23/2013. nces.		The above described repai this repair station under W	r or maintenance is approved for re ork Order #3TL9A.	eturn to service. Pertinent details of this	repair/maintenance are on file
THIS AIRC	D FOR RETURN TO S	ERVICE FOR TH	E WORK PERFORMED. P	WITH CURRENT FAA REGULAT ERTINENT DETAILS ARE ON FILE	IONS AND IS UNDER THIS	I certify that this aircraft/a airworthy condition/safe for	appliance has been inspected in according to the strength of t	cordance with inspections referenced aborspection are on file at this repair station	ove and was determined to be under Work Order #3TL9A.
	INSPECTOR:		EPAIR STATION WORK O	RDER: 04029962.	3	Signed:	DUNC QI 12	AN 24 for Duncan Avionic	es; Englewood, CO 80112
	EXECUTIVE A	IRCRAFT MAINT E. GREENWAY	ENANCE FAA APPROVE ROAD SCOTTSDALE, AZ	D REPAIR STATION E27R1730 85260 480-991-0900					Total Landings: 539
Carry Forward									



Registration #: N552TC Hobbs: 2190.3

Total Aircraft Time: 2190.3 Total Engine Time: 2190.3 Total Propeller Time: 951.8

Total Aircraft Cycles: 1561 Total Engine Cycles: 1561 Total Propeller TSOH: 951.8

VCCS Hours: 539.0

Work Order #: 2815

Date: 06/26/2015

PC-12/45, S/N 443

-Performed an Annual Inspection using the Pilatus PC-12/45 checklist and FAR 43 Appendix D.

-Airworthiness Directive compliance checked through Revision 2015-12.

-Serviced aircraft oxygen referencing Pilatus PC-12 AMM Section DMC-12-A-12-10-05-00A-902A-A.

-Downloaded trend monitoring information and forwarded to The Trend Group.

-Complied with continued airworthiness inspection requirements for installed equipment as required.

Aircraft Airwor niness Directives and Service Bulletins

Complied with Pilatus flap control system current surges by performing wing flap actuator backlash check referencing Pilatus PC-l'AMM Section DMC-12-A-27-50-03-00A-313A-A and AD 2001-25-10 with results as follows:

	- /	P/N	S/N	Previous Backlash	Current Backlash	Total Hours	Total Lndgs
	Rt Otte	978.73.20.309	1409	.007	.008	2190.3	1561
	Rt Inbd	978.73.20.309	0131	.008	.009	2190.3	1561
	Li Otbd	978.73.20.309	1044	.007	.008	2190.3	1561
d	Lt Inbd	978.73.20.309	1397	.009	.009	2190.3	1561

Aircraft total landings 1561. Recurring inspection due each 600 hours or annual inspection. Next inspection due at 2790.3 hours or 06/2016.

-Complied with Pilatus AD 2014-22-01 by inserting Structural and Component Limitations/Airworthiness Limitations Document inche AMM. The flap drive arm insitu inspection has been complied with at this time.

-Complied with AD 2015-08-07 Flight Crew Oxygen Mask Regulator. Not applicable by date of manufacture.

-Complied with Pilatus mandatory SB 72-005 (329) - Inspection of Fuel Nozzle Tips.

-Complied with Pilatus SB 34-041 (324) - Improvement to the Pitot Pipe Line (L/H Wing Pitot System). Removed the plastic pitothe located in the L/H outer wing and replaced with an aluminum pipe, P/N 500.50.12.352, referencing the service bulletin instruction -Complied with Pilatus SB 52-007 (323) R2 Part A, B, & C. The cabin door was NDT inspected by NDE Services. See Eddy Cum

Certification Report dated 6/23/15 placed in the aircraft records. Eddy current inspection completed on passenger door S/N 332. Alttings determined to be manufactured from AA 2124 T851 material and no repetitive inspections are required.

-Complied with Pilatus SB 21 012 (307) - Part is not applicable per serial number 2176. -Complied with Pilatus SB 24-035 (300) - Protection of the Engine Bay Electrical Harness referencing the service bulletin instructic Not applicable per the aircraft serial number.

-Complied with Pilatus SB 11-005 (287) - Replace the Fuel Filler Placard referencing the service bulletin instructions.

P:1 of 3

AIRFRAME (continued) N552TC Work Order # 2815 ----Aircraft Scheduled Maintenance and Inspections

-Complied with 6 month application of AgeMaster to surface de-icers referencing Pilatus PC-12 AMM Section DMC-12-A-05-10-20-00A-281A-A

-Complied with 1 year/3000 hour horizontal stabilizer trim runaway aural warning functional test referencing Pilatus PC-12 AMM

Section DMC-12-A-27-40-00-00A-903A-A. Functional test satisfactory. -Complied with 12 month corrosion and preventative compound condition inspection referencing Pilatus PC-12 AMM Section

DMC-12-A-20-40-00-00A-901A-A. No discrepancies noted. -Complied with 1 year fire extinguisher contents check referencing Pilatus PC-12 AMM Section DMC-12-A-26-20-00-00A-040A-A. No

discrepancies noted.

-Complied with 1 year/150 hour flap actuator lubrication service referencing Pilatus PC-12 AMM Section DMC-12-A-27-50-00-00A-902A-A.

-Complied with 1 year/150 hour inboard flap drive arm in-situ inspection/check referencing Pilatus PC-12 AMM Section DMC-12-A-27-51-00-00A-313A-A. No discrepancies noted.

-Complied with 12 month lubrication of main and nose wheel bearings referencing Pilatus PC-12 AMM Section DMC-12-A-12-20-06-00A-902A-A

-Complied with 1 year/2000 hour MLG axle and bushing inspection referencing Pilatus PC-12 AMM Section DMC-12-A-32-10-00-00A-313A-A. No discrepancies noted.

-Complied with 1 year/500 hour engine torque and stick pusher torque transducer electrical connectors cleaning referencing Pilatus PC-12

-Complied with 1 year/3000 hour propeller feathering microswitch functional tests referencing Pilatus PC-12 AMM Section DMC-12-A-76-10-02-00A-903A-A. Functional test satisfactory.

-Complied with 200 hour starter/generator brush check referencing Pilatus PC-12 AMM Section DMC-12-A-24-30-01-00A-313A-A.

-Complied with 1 year VCCS cleaning/inspection referencing Enviro Systems CMM 21-00-23. No discrepancies noted.

-Complied with I year/1000 hour autopilot clutch torque check referencing Pilatus PC-12 AMM Section DMC-12-A-22-10-00-00A-313A-A. No discrepancies noted.

-Complied with 1 year crew oxygen mask operational test and cleaning referencing Zodiac Aerospace MC10 Series Operating and Maintenance Instructions 4NUT0045A. Operational test satisfactory.

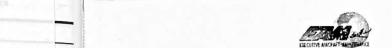
-Complied with 3 year passenger oxygen mask inspection referencing Pilatus PC-12 AMM Section DMC-12-A-35-20-00-00A-313A-A. -Complied with 1000 hour generator 2 drive assembly housing inspection referencing Pilatus PC-12 AMM Section DMC-12-A-24-30-04-00A-313A-A. No discrepancies noted.

-Complied with 400 hour generator 2 brush inspection referencing Pilatus PC-12 AMM Section DMC-12-A-24-30-02-00A-313A-A. No

Carry Forward

Registration No._ No. 31802094 51612125 arings with en, IAW Pil 500 hour vapor us PC12 MM 21oved and repla 500 hour 1-00A-313 200 hour 24-30-01

A:---- & T:---



AIRFRAME Pilatus PC12/45

3/18/2015 S/N: 443 -

N552TC

AC Hobbs: 2125.3

Removed copilot GDU 620 fan plate assembly P/N K00-00683-00 NSN and installed new fan kit P/N K00-00683-00 NSN

IAW Garmin manual 190-00601-10, Rev. 1, 2/2014. Operational check satisfactory. 2. Repaired # 2 GDU 620 to EIS wire as required. Ground operational check satisfactory IAW Garmin manual 190-00601-06.

Rev. L. 10/2013. All work accomplished IAW Doc. 02049 MM Rev. 29 and above references.

THIS AIRCRAFT WAS INSPECTED AND/OR REPAIRED IN ACCORDANCE WITH CURRENT FAA REGULATIONS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTIDENT, DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION WORK ORDER: 94029916.

INSPECTOR:

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EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260 480-991-0900

AIRCRAFT LOG Aircraft Time Recorder Date Maintenance This Date Total Reading Brought Forward AIRFRAME N552TC Pilatus PC12/45 AC Hobbs: 2370.6 3/1/2016 S/N: 443 Installed Garmin FlightStream 210 P/N 011-03257-40 S/N 3RE003915 IAW Garmin manual 190-01007-A3, Rev. 9, 2/2016. Weight change negligible. Ground operational check satisfactory. All work accomplished IAW Doc. 02049 MM Rev. 31 and above references. THIS AIRCRAFT WAS INSPECTED AND/OR REPAIRED IN ACCORDANCE WITH CURRENT FAA REGULATIONS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION WORK ORDER: 4032026 INSPECTOR: EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDALE, AZ 85260 480-991-0900 Total to Carry Forward

AIRFRAME LOGBOOK ENTRY

DATE: April 21, 2016 A/C REG. #: N552TC S/N: 443 MODEL: PC 12/45

TOTAL TIME: 2397.9 TOTAL CYCLES: 1722.0 WORK ORDER: 12

DESCRIPTION OF WORK PERFORMED:

-Performed an Annual Inspection using the Pilatus PC-12/45 checklist and FAR 43 Appendix D as a guide.

-Airworthiness Directive Compliance Checked Through Revision 2016-08. And Found To Be Current At This Time. -Complied With Continued Airworthiness Inspection Requirements For Installed Equipment As Required.

Aircraft Airworthiness Directives and Service Bulletins Complied With

- AD 2015-19-15 Universal Joint Replacement Does Not Apply By The Aircraft Serial Number And Has Not Had Universals Replaced. -Complied With AD 2001-25-10 and Pilatus Flap Control System Current Surges By Performing Wing Flap Actuator Backlash Check

Referencing Pilatus PC-12 AMM Section DMC-12-A-27-50-03-00A-313A-A With Results As Follows: Total Hours Total Lndgs Previous Backlash Current Backlash S/N 1722 2397.9 .008 Rt Otbd 978.73.20.309 1722 2397.9 .009 009 0131 Rt Inbd 978.73.20.309

1722 2397.9 008 008 978 73 20 309 1044 Lt Otbd 2397 9 1722 009 1387 .009 978.73.20.309 I t Inbd Aircraft total landings 1722. Recurring inspection due each 600 hours or annual inspection. Next inspection due at 2997.9 hours or 04/2017.

Aircraft Scheduled Maintenance and Inspections Complied With:

-1 year/3000 Hour Horizontal Stabilizer Trim Runaway Aural Warning Operational And Functional Tests. No Defects Noted. Ref AMM 12-- A-27-40-00-00a-903A-A

-1 Year/3000 Hour Propeller Feathering Microswitch Functional Test. Ref Pilatus Maintenance Manual DMC-12-A-76-10-02-00A-903A-A, No Defects Noted.

-1 Year/2000 Hour Main Landing Gear Axle And Bushing Inspection. No Defects Noted. Ref AMM 12-A-32-10-00-00A-313A-A

-1 Year/1000 Hour Autopilot Clutch Torque Check Complied With Ref DMC-12-A-22-10-00-00A-313A-A. No Discrepancies Noted.

- - 1 Year/500 Hour Engine Torque And Stick Pusher Torque Transducer Electrical Connectors Cleaning Ref AMM-12-A-71-00-03-00A-250A-A

-1 Year/150 Hour Inboard Flap Drive Arm In-Situ Inspection/Check Ref AMM-12-A-27-51-00-00A-313A-A.

-1 Year/150 Hour Flap Actuator Lubrication Service Ref AMM-12-A-27-50-03-00A-313B-A.

-1 Year VCCS Inspection/Cleaning Ref AMM-12-A-21-50-03-00A-313A-A. Cooling to 55 Deg F.

-1 Year Fire Extinguisher Contents Check. Ref Pilatus Maintenance Manual DMC-12-A-26-20-00-00A-040A-A. No defects noted.

-1 Year ELT Inspection Referencing FAR 91.207 (d), No Discrepancies Noted By Executive Aircraft Maintenance ref WO 4032080.

AIRFRAME LOGBOOK ENTRY	2 of 2
DATE: April 21, 2016	

A/C REG. #: N552TC S/N: 443 MODEL: PC 12/45

TOTAL TIME: 2397.9 TOTAL CYCLES: 1722.0 WORK ORDER: 12

Aircraft Scheduled Maintenance and Inspections Complied With: (Continued)

- -1 Year Corrosion And Preventative Compound Condition Inspection. Ref Pilatus Maintenance Manual DMC-12-A-20-40-00-00A-901A-A No Defects Noted
- -1 Year Lubrication Of Main And Nose Landing Gear Wheel Bearing Ref Pilatus Maintenance Manual DMC-12-A-12-20-06-00A-902A-A
- -600 Hour/6 Month Lead Acid Battery Capacity Test, No Defects Noted, Ref Concorde Battery CMM 24-30-18.
- -6 Month Application Of AgeMaster To Surface Deicers By West Coast Wash Station, PO16-00017.2. -300 Hour/400 Landings NLG Drag Link Right Part Inspection Complied With By Canyon State Inspection On CSI Job# AC16-267. PO16-
- 00017
- -200 Hour Starter/Generator Brush Check Ref Pilatus Maintenance Manual DMC-12-A-24-30-01-00A-313A-A.
- -100 Hour MT-V Propeller Inspection Ref MT-V CMM 61-00-83
- -Downloaded Trend Monitoring Information Ref Logic EIS Trend Download Manual, Issue 1

Aircraft Maintenance Actions

-Serviced Aircraft Oxygen. No Leaks Or Defects Noted. Ref Pilatus Maintenance Manual DMC-12-A-12-10-05-00A-902A-A -Aircraft Paint Touched Up And Chip Repaired By Passenger Door. Work Performed By Horizon Aircraft Finishes on PO16-00019.3 -Replaced NLG Torque Link Lubrication Bolt And Bushings, No Other Defects Noted. Ref CMM 32-20-05

-Replaced Nose Wheel Bearing Races. No Defects Noted. Ref Goodrich CMM 3-1501 (32-47-72). -Replaced Elevator To Vertical Stabilizer Bonding Straps. No Defects Noted. Ref AMM DMC-12-A-55-20-01-00A-920A-A. -Replaced 3 Cabin/Passenger Oxygen Masks P/NP75710. Ref.DMC-12-A-35-20-01-00A-920A-A

-Replaced Forward Most Oxygen Outlet, Forward Of Cargo Door. Leak check Good. Ref AMM-12-A-35-20-01-00A-920A-A.

1 Certify That This Aircraft Has Been Inspected In Accordance With An ANNUAL Inspection And Was Determined To Be In Airworthy Condition. All Details Of Work Performed Is On File At This Facility.

SIGNED: Bell Lon DATE: April 21, 2016

TEMPUS JET CENTERS IV, 8014 E. McClain Dr. Ste.110, SCOTTSDALE AZ 85260

NOTES



AIRERAME Pilatus PC12/45

4/14/20 5/14: 4

AC Hobbs: 2397.9

NESSTC.

- C/W 12 month Inspection of ELT IAW FAR 91.207 (d) with no discrepancies noted. Baftery Expires: 11/2020.
- Removed roll servo actuator P/N 055-0056-56, S/N 5161, yaw servo actuator P/N 065-0055-80, S/N 5125 and pitch ser actuator P/N 065-0056-59, S/N M4540. Customer to re-install removed actuators, no further action.
- All work accomplished IAW Doc. 02049 MM Rev. 31 and above references

THIS AIRCRAFT WAS INSPECTED AND/OR REPAIRED IN ACCORDANCE WITH CURRED THAN REGULATIONS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED, PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATIC

WORK ORDER: 4032080.

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTS DALE, AZ 85260 480-991-0900

DATE: April 21, 2016

A/C REG. #: N552TC S/N: 443 MODEL: PC 12/45

TOTAL TIME: 2397.9 TOTAL CYCLES: 1722.0 WORK ORDER: 12

Addendum to Annual Inspection Entry Dated 04/21/2016 DESCRIPTION OF WORK PERFORMED:

-Removed Propeller P/N: MTV-27-1-N-C-F-R(P)/CFR260-65a, Hub S/N: 100793 TT 1147.0 Installed Propeller Overhauled by MT-Propeller WO# 14-211MT and assembled by Warner Propeller WO# 12016 Prop P/N MTV-27-1-N-C-F-R(P)/CFR260-65a Hub S/N 100252 Blade S/N's 1)ADE-46227 2) ADE-46228 3) ADE-46229 4) ADE-46230 5) ADE-46231. Hub TSN 428.6 Hrs. Blades TSN 1957.8 Hrs. Propeller TSO 0.0 hrs. Operational Check Completed With No Defects Noted. Ref MT-Propeller ATA 61-00-83-E-1803 Rev 23 and DMC-12-A-71-00-00-000A-903A-A. Verified Propeller Balance Below .2 ips Ref DMC-12-A-61-00-00-00A-903A-A.

I certify that this aircraft has been inspected in accordance with an Annual and was determined to be in airworthy condition. All details of work performed is on file at this facility.

SIGNED: DATE: April 21, 2016

(Bill Willingham A&P 393768866 IA)

TEMPUS JET CENTERS IV, 8014 E. McClain Dr. Ste. 110, Scottsdale AZ 85260

NOTES



Registration #: N552TC Hobbs: 2459.3 Total Aircraft Time: 2459.3 Total Engine Time: 2459.3 Total Propeller Time: N/A

Work Order #: 3131

Total Aircraft Cycles: Total Engine Cycles: N/A Total Propeller TSOH: N/A

Date:6/16/2016

PC-12/, S/N 443

Performed Main Landing Gear Forward and Aft Attachment bolt & Main Landing Gear Actuator Bolt Inspection. Referencing 12A-32-10-00-00A-310A-A. Replaced right front MLG attach bolt P/N932.35.14.579. Due to corrosion. All Other bolts did not have any corrosion and were reinstalled with corrosion inhibiting compound.

The Aircraft was repaired and/or inspected in accordance with current Pilatus PC-12 Amm Rev ision 25This aircraft was found to be airworthy and is approved for return to service. Pertinent details are on the at this facility under the work order referenced below.

For Tempus Aircraft Sales & Service: 1 14 14823

WO# A3135

CONTINUED IN LOOKBOOK #2



AIRFRAME Pilatus PC12/45

N552TC

AC Hobbs: 2443.7

5/31/2016 S/N: 443

- 1. Removed #1 GDU-620. Replaced fan assembly with new P/N K00-006803-10 in reference to 190-00601-10. Re-installed GDU-620 and ops checked in reference to 190-00601-06 Rev. N with no defects noted.
- 2. Work performed in reference to Doc. 02049 MM Rev. 31 and above references.

THIS AIRCRAFT WAS INSPECTED AND/OR REPAIRED IN ACCORDANCE WITH CURRENT FAA REGULATIONS AND IS APPROVED FOR RETURN TO SERVICE FOR THE WORK PERFORMED. PERTINENT DETAILS ARE ON FILE UNDER THIS FAA APPROVED REPAIR STATION WORK ORDER: 4032681.

INSPECTOR:

EXECUTIVE AIRCRAFT MAINTENANCE FAA APPROVED REPAIR STATION E27R1730 7920 E. GREENWAY ROAD SCOTTSDA! E, AZ 85260 480-991-0900